# Connections in Precast Concrete Structures-Strength of Corbels 

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## SYNOPSIS

This paper describes a project directed toward development of design criteria for reinforced concrete corbels. Part 1 contains these design criteria, together with design aids and design examples. Part 2 describes the tests on which the proposed criteria are based, involving 124 corbels subjected to vertical loads only and 71 corbels subjected to combined vertical and horizontal loads. Part 3 contains the discussion and analysis of the experimental data and the derivation of the design equations. Detailed test data are given in an appendix.

## introduction

A series of investigations of connections in precast concrete structures is in progress at the Research and Development Laboratories of the Portland Cement Association. The three previous papers in this series, collectively entitled "Connections in Precast Concrete Structures," have been concerned with the strength and behavior of continuity connections in double-tee floor construction ${ }^{1}$, with the bearing strength of column heads supporting precast beams ${ }^{2}$, and with the strength and behavior of scarf joints in beams and

[^0]columns ${ }^{3}$. This paper deals with the development of design criteria for the strength of corbels which protrude from the face of a column.

## PART I-DESIGN OF CORBELS

## Background

Corbels projecting from the faces of columns are used extensively in precast concrete construction to support primary beams and girders. Typical applications of corbels may be found in the Prestressed Concrete Institute manual of connection details ${ }^{4}$.
Until recent years little research had been available on the strength of corbels. In the United States it has been customary to design them as short cantilevers, using the flexural and shear design equations derived for beams of more normal propor-
tions. Since the assumptions made in deriving these equations are not valid for deep beams, it is not surprising that corbel brackets designed by these equations can have varying safety factors. The tests described in Part 2 of this paper show that design on this basis will lead to questionably safe designs when the amount of tension reinforcement exceeds about one percent, and also if shear reinforcement is necessary and is provided in the form of vertical stirrups. In addition, corbels have in general been designed for vertical loads only, although horizontal forces caused by restrained creep, shrinkage, and temperature deformations of the beams supported by the corbels are often important indeed. Tests described in Part 2 of this paper have shown that such horizontal forces can substantially reduce the vertical load-carrying capacity of corbels. This effect has also been evidenced in the field where some corbels carrying light vertical loads were damaged by horizontal restraint forces.

In Europe the design of corbels has been based mainly on the investigations of Rausch ${ }^{5,6}$. These design procedures involve the "straight-line" method of design for flexure, and the provision for bent bars to resist all shear forces.

In 1961, Niedenhoff ${ }^{7}$ suggested that a corbel acts essentially as a simple truss composed of two members: a horizontal tension member, i.e. the tension reinforcement, and an inclined concrete compression strut. On the basis of an experimental investigation, Niedenhoff proposed that the depth of the equivalent truss be taken as 0.8 times the total depth of the corbel. These assumptions form the basis of Niedenhoff's working load design proce-
dure.
A series of tests conducted at the University of Illinois ${ }^{8,9,10,11}$ involved the strength of deep beams. A deep beam, loaded by a concentrated load at midspan and supported by concentrated reactions at the ends, acts essentially as a double corbel protruding from opposite faces of a column. However, the number of specimens tested under concentrated loads was not sufficient to lead to design procedures for corbels. These tests, together with recent tests of short cantilevers made at the University of Texas ${ }^{12}$, will be referred to later.

The tests recently carried out in the PCA Structural Laboratory, and reported in this paper, have been specifically concerned with corbels in which the ratio of the shear span to the effective depth of the bracket at the column face was less than unity. One hundred ninety-five corbels were tested, of which 124 were subject to vertical load only and 71 to combined vertical and horizontal loads. The variables included in the tests were: size and shape of corbel, amount of main tension reinforcement and its detailing, concrete strength, amount of stirrups, ratio of shear span to effective depth, and the ratio of the horizontal force to. the vertical force.

The design criteria set out below are based on a study of the results of these tests; they have also been checked against the results obtained from the tests at the Universities of Illinois ${ }^{8,9,10,11}$ and Texas ${ }^{12}$. In the development of such design criteria, numerous plots and numerical computations were made to compare observed performance with various empirical expressions. Considerable use was made of electronic computation to arrive at suitable ultimate
strength design equations.

## Proposed Criteria for the Design of Corbels

1. Notation
$A_{s}=$ area of tension reinforcement, in. ${ }^{2}$
$A_{v}=$ total area of horizontal closed stirrups, in. ${ }^{2}$
$a \quad=$ shear span, i.e. distance from column face to resultant of vertical load, in.
$b \quad=$ width of corbel, in.
$d \quad=$ effective depth of corbel measured at column face, in.
$f_{c}^{\prime}=$ concrete cylinder strength, psi
$\sqrt{f_{c}^{\prime}}=$ relationship expressed in psi , so that $\sqrt{f_{c}^{\prime}}=60 \mathrm{psi}$ for $f_{c}^{\prime}=3600 \mathrm{psi}$
$H / V=$ ratio of horizontal load to vertical load
$p=$ reinforcement ratio at column face,
$p=\frac{A_{s}+A_{v}}{b d}$ when $H / V=$ 0 , i.e. vertical loads only, $p=\frac{A_{s}}{b d}$ when $H / V$ does not equal zero, i.e. combined vertical and horizontal loads
$v_{u}=$ nominal shear stress at ultimate strength, psi, $v_{u}=\frac{V_{u}}{b d}$
$V_{u}=$ vertical load at ultimate strength, i.e. shear at ultimate strength, Ib
$\phi=$ capacity reduction factor
2. Scope
(a) These provisions apply to corbel brackets having a shear span to depth ratio, $a / d$, of less than unity.
(b) Provisions of the ACI Building Code (ACI 318-63) not in conflict
with the provisions of these proposed criteria should be considered applicable to the design of corbels.

## 3. Safety Provisions and Design Loads

(a) Strength should be computed in accordance with the provisions of section 4.
(b) The coefficient $\phi$ should be 0.85 .
(c) The strength capacities of corbels so computed should be at least equal to the total effects of the design loads required by Section $3(\mathrm{~d})$.
(d) The design loads to be used in the design of corbels should equal the design loads specified in Section 1506 of the ACI Building Code (ACI 318-63), multiplied by $4 / 3$.

## 4. Strength Computations

(a) When special provisions are made so that a corbel is subject to vertical loads only, the ultimate design load capacity may be calculated by:

$$
\begin{align*}
& V_{u}=\phi\left[6.5 b d \sqrt{f_{c}^{\prime}}\left(1-0.5^{d / a}\right)\right. \\
& \left.(1000 p)^{1 / 3}\right] \tag{1}
\end{align*}
$$

where $p=\left(A_{s}+A_{v}\right) / b d$ does not exceed 0.02 , and $\mathrm{A}_{v}$ does not exceed $\mathrm{A}_{s}$.
(b) In all other cases the ultimate design load capacity may be calculated by:

$$
\begin{align*}
& V_{u}=\phi\left[6.5 b d \sqrt{f_{e}^{\prime}}\left(1-0.5^{d / a}\right)\right. \\
& \left.\frac{(1000 p))^{(1 / 3+0.4 H / V)}}{10^{0.8 H / V}}\right] \tag{2}
\end{align*}
$$

where $p=A_{s} / b d$ does not exceed 0.013 .

## 5. Minimum reinforcement

(a) The amount of tension reinforcement $A_{s}$ should be not less than $0.004 b d$.
(b) Closed horizontal stirrups should be provided having a total


Fig. 1-Recommended Corbel Details
cross section $\mathrm{A}_{v}$ not less than $0.5 \mathrm{~A}_{s}$.

## 6. Detailing of Corbels*

(a) The tension reinforcement should be anchored as close to the outer face of the corbel as cover requirements permit, by welding a cross-bar to the ends of the tension reinforcing bars. The size of the cross bar should be at least equal to the maximum size for bar used as tension reinforcement.
(b) The closed horizontal stirrups should be distributed over the up-

[^1]per two thirds of the effective depth at the column face.
(c) The total depth of a corbel under the outer edge of a bearing plate resting on the corbel should be not less than half the total depth of the corbel at the face of the column.
(d) The outer edge of a bearing plate resting on a corbel should be placed not closer than 2 inches to the outer edge of the corbel.
(e) When corbels are designed to resist horizontal forces, steel bearing plates welded to the tension reinforcement should be used to transfer the horizontal forces directly to the tension reinforcement.

## 7. Bearing Stresses

(a) The bearing stresses at ultimate strength beneath a bearing plate resting on a corbel should be not more than $0.5 f_{c}^{\prime}$.

## Discussion of Proposed Design Criteria

## Safety Provisions and Design Loads

The proposed safety provision and design loads are in agreement with the philosophy concerning safety provisions and design loads of Part IV-B, Ultimate Strength Design, of the ACI Building Code (ACI 318-63). Since a corbel is primarily a shear transfer device, and since its ultimate strength is governed by shear strength, it is considered appropriate to use the value $\phi=0.85$ specified in ACI 318-63 for ultimate strength governed by shear and diagonal tension.

The design loads specified for corbels are made one third greater than those specified for the design of members in ACI 318-63 for two reasons. First, in corbels having less than about one percent of tension reinforcement, yield of the reinforcement occurs before the ultimate strength of the corbel is developed. The ratio of the load at which yield occurs to the ultimate load can vary
between $\%$ and 1 . The load factors proposed will provide an adequate factor of safety against yield of the reinforcement, thus insuring serviceability of the corbels under moderate overloads. Second, it is considered good practice that the strength of a precast concrete structure should be governed by the strength of the members and not by the strength of the connections between members. Since a corbel forms part of the connection between a beam and a column it should be made stronger than either the beam or the column. Use of the proposed design loads will assure this.

## Strength Computations

The equations for ultimate strength presented in Section 4 and Part 3 are based on a study of the results of tests of 195 corbels carried out at the PCA Structural Laboratory. Eq (2) reduces to Eq. (1) when $H / V$ is zero. However, the different definitions of reinforcement ratio $p$ in Eqs. (1) and (2) should be noted. Whereas stirrups make a considerable and consistent contribution to the strength of a corbel subject to vertical load only, their contribution to the strength of a corbel subject to combined vertical and horizontal loads is smaller and more variable. It is therefore considered sounder for the present not

Table 1-Comparison of Test and Calculated Strengths

| Source | Type of Specimen | Number of Specimens | H/V | Average $V_{w}$ test <br> $V_{u}$ calc | Standard Deviation |
| :---: | :---: | :---: | :---: | :---: | :---: |
| PCA | Corbels without stirrups | 78 | 0 | 1.02 | 0.119 |
| PCA | Corbels with stirrups | 10 | 0 | 1.11 | 0.084 |
| PCA | Corbels without stirrups | 25 | 1/2 | 1.05 | 0.132 |
| PCA | Corbels without stirrups | 21 | 1 | 1.21 | 0.216 |
| $\underset{\text { PCA }}{\text { Pf }{ }^{\text {8,0,11 }}}$ | Corbels with stirrups | 4 | 1 | 1.42 | - |
| U of $\mathrm{I}^{\text {P,10,11 }}$ | Deep beams | 23 | 0 | 1.01 | 0.134 |
| $\underset{\mathrm{U}}{\mathrm{U}}$ of of $\mathrm{T}^{12}$ | Beams with $a / d=1.33$ Short cantilevers $\bar{a} / d<1.10$ | 14 | 0 | 1.14 | 0.168 |
| U of $\mathrm{T}^{\text {I2 }}$ | Short cantilevers $a / d<1.10$ | 6 | 0 | 1.03 | 0.066 |

to rely on their contribution when designing a corbel subject to combined loading.
Eqs. (1) and (2) have been used to calculate the strengths of 181 members tested at PCA, the University of Illinois, and the University of Texas. Tests involving local failures resulting from inadequate reinforcing details were excluded. A summary of the results of this application of the proposed equations is set out in Table 1. In these calculations, $\phi$ was taken equal to 1.0 , since accurate values of material properties and of dimensions were known.
The application of these equations is simplified considerably by the use of design aids which are presented following this discussion.

## Minimum Reinforcement

The minimum amount of tension reinforcement is specified to insure against too rapid opening of cracks after first cracking. The lower the amount of tension reinforcement, the lower is the ratio of load at yield of tension reinforcement to ultimate load.

Closed horizontal stirrups are required in all corbels to eliminate the possibility of a sudden explosivetype failure of the corbel, which can occur in a corbel without stirrups.

## Detailing of Corbels

The correct detailing of corbels is fully as important as the over-all design of the reinforcement. Almost invariably, distress of corbels in the field can be traced to poor detailing. If the tension reinforcement is not effectively anchored close to the outer face of the corbel, the full strength potential of the reinforce-
ment cannot be developed and failure will occur at a lower load than indicated by Eqs. (1) and (2). The recommended form of anchorage using a bar welded across the ends of the tension reinforcement is shown in Fig. 1. A frequently used detail for the main tension reinforcement is shown in Fig. 2(a). However, in order to conform to Section 801 of the ACI Building Code which specifies minimum bend radii for reinforcing bars, the bars are actually bent as shown in Fig. 2(b). Failure has then been observed, both in the field and the laboratory, to occur on the surface indicated in Fig. 2(b), the tension reinforcement being bypassed completely. Welding of the bearing plate to the main reinforcement when horizontal forces act is specified to eliminate the possibility of a local failure of the concrete between the bearing plate and the reinforcement.

The horizontal stirrups are located so that they will be as effective as possible, both from consideration of ultimate strength and for control of diagonal cracks. A suitable spacing of stirrups, $s$, is given by

$$
s=\frac{2}{3}\left(\frac{d}{n+1}\right)
$$

where $n$ is the number of stirrups used. The stirrups should be placed in the corbel beginning at a distance $s$ from the tension reinforcement. Horizontal stirrups are used rather than vertical stirrups because of the steep inclination of the diagonal cracks. These cracks can in some cases be almost vertical.

The limiting proportions of a corbel, and the limiting location of the bearing plate, are both recommended to insure against local failures of the concrete before the po-


Fig. 2-Corbel Details
tential strength of the corbel has been developed. If the outer face of the corbel is made too shallow, the principal diagonal crack will take a course as shown in Fig. 2(c), and will intercept the sloping face of the corbel. resulting in instantaneous failure. If the outer face is sufficiently deep, however, the principal diagonal crack will take a course as shown in Fig. 2(d). In this case a diagonal concrete compression strut is formed as indicated, and a further increase in load may be possible after formation of the crack. Location of the bearing plate too close
to the outer face of the corbel can result in a bearing failure beneath the plate at relatively low intensities of stress. This is particularly the case if the load on the bearing plate becomes eccentric. It is essential to insure that rotation of the end of a beam due to deflection under load shall not result in the beam bearing on the outer edge of the corbel.

## Bearing Stresses

Use of the maximum bearing stress of $0.5 f_{c}^{\prime}$ is contingent upon compliance with the requirements of Section 6(d). Bearing failures were ex-
perienced at stresses lower than $0.5 f_{c}^{\prime}$ in corbels loaded through bearing plates located closer to the outer face than two inches.

## Design Aids and Design Examples

## Design Aids

Design aids have been prepared to facilitate the use of Eqs. (1) and (2).

Eq. (1) may be written:

$$
\begin{equation*}
V_{u}=\phi b d \sqrt{f_{e}^{\prime}} F_{1} F_{2} \tag{1a}
\end{equation*}
$$

where $\quad F_{1}=6.5\left(1-0.5^{d / a}\right)$, and

$$
F_{2}=(1000 p)^{1 / 3}
$$

Values of $F_{1}$ and $F_{2}$ are listed in Tables 2 and 3.

Similarly, Eq. (2) may be written:

$$
\begin{equation*}
V_{u}=\phi b d \sqrt{f_{c}^{\prime}} F_{1} F_{3} \tag{2a}
\end{equation*}
$$

where $\mathrm{F}_{3}=\frac{(1000 p)^{(1 / 3+0.4 H / V)}}{(10)^{0.8 H / V}}$
Values of $F_{3}$ are listed in Table 4. Using Eqs. (la) or (2a), and Tables 2,3 , and $4, V_{u}$ may be readily evaluated for given values of $b, d, f_{c}^{\prime}$ $p, a / d$, and $H / V$. The use of the tables is illustrated in the following examples.

Since both $p$ and $a / d$ can be varied independently, design of a corbel must be by successive trials. This process is simplified by use of the design chart given in Fig. 3. It is proposed that corbels be designed by successive trials using the design chart, and that the strength of the final design be checked using either Eq. (la) or (2a), whichever is appropriate. Use of the chart and equations in this manner is illustrated in the examples.

## Example 1

A typical interior corbel shown in Fig. $4(\mathrm{a})$ projects from a $14 \times 14$-in.
square tied column. It supports a 50 -ft span prestressed girder carrying a live load of $1500 \mathrm{lb} / \mathrm{ft}$ and a dead load of $960 \mathrm{lb} / \mathrm{ft}$. Design the corbel for the vertical reaction from the girder, assuming that suitable bearings are provided to eliminate horizontal restraint forces, and that the corbel does not have to resist wind or earthquake forces. Intermediate grade reinforcement is used and $f_{c}^{\prime}=5000$ psi. Tolerance gap between beam end and column face is one inch.

- Design Loads.

Dead load reaction $=24 \mathrm{kips}$
Live load reaction $=37.5 \mathrm{kips}$ Ultimate design load,

$$
\begin{aligned}
V_{u} & =4 \\
3 & -(1.5 D+1.8 L) \\
& =2.0 D+2.4 L \\
& =2.0(24)+2.4(37.5)
\end{aligned}
$$

$$
V_{u}=138 \mathrm{kips}
$$

- Determine shear span " $a$ ". $a \cong 2$ (tolerance gap between beam and column) $+1 / 2$ (bearing plate width)

$$
\begin{aligned}
& \text { Bearing plate width }=\frac{V_{u}}{b\left(f_{e}^{\prime} / 2\right)} \\
& \quad=\frac{138,000}{14 \times 2500}=3.9 \mathrm{in} ., \text { say } 4 \mathrm{in} . \\
& a=2(1)+4 / 2=4 \mathrm{in} .
\end{aligned}
$$

- Estimate depth $d$.
$a^{\prime} / d$ is generally between 0.15 and 0.4 ; assume $a / d=0.3$, hence $d=$ 13.3 in.
- Determine $v_{u}=V_{u} / b d$.

$$
v_{u}=\frac{138,000}{14 \times 13.3}=741 \mathrm{psi}
$$

- Find required $p$ from design chart. Enter chart at $v_{u}=741 \mathrm{psi}$, proceed horizontally to $f_{r}^{\prime}=5000 \mathrm{psi}$, vertically to $a / d=0.3$, horizon-

Table 2-Values of $F_{1}=6.5\left(1-0.5^{d / a}\right)$

| $a / d$ | 0.00 | 0.01 | 0.02 | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 | 0.08 | 0.09 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 0.0 | 6.50 | 6.50 | 6.50 | 6.50 | 6.50 | 6.50 | 6.50 | 6.50 | 6.50 | 6.50 |
| 0.1 | 6.49 | 6.49 | 6.48 | 6.47 | 6.45 | 6.44 | 6.41 | 6.39 | 6.36 | 6.33 |
| 0.2 | 6.30 | 6.26 | 6.22 | 6.18 | 6.14 | 6.09 | 6.05 | 6.00 | 5.95 | 5.90 |
| 0.3 | 5.85 | 5.80 | 5.75 | 5.70 | 5.65 | 5.60 | 5.55 | 5.50 | 5.45 | 5.40 |
| 0.4 | 5.35 | 5.30 | 5.25 | 5.20 | 5.15 | 5.10 | 5.06 | 5.01 | 4.97 | 4.92 |
| 0.5 | 4.87 | 4.83 | 4.79 | 4.74 | 4.70 | 4.66 | 4.61 | 4.57 | 4.53 | 4.49 |
| 0.6 | 4.45 | 4.41 | 4.37 | 4.34 | 4.30 | 4.26 | 4.22 | 4.19 | 4.15 | 4.12 |
| 0.7 | 4.08 | 4.05 | 4.02 | 3.98 | 3.95 | 3.92 | 3.89 | 3.86 | 3.83 | 3.80 |
| 0.8 | 3.77 | 3.74 | 3.71 | 3.68 | 3.65 | 3.62 | 3.60 | 3.57 | 3.54 | 3.52 |
| 0.9 | 3.49 | 3.46 | 3.44 | 3.42 | 3.39 | 3.37 | 3.34 | 3.32 | 3.30 | 3.27 |

Table 3-Values of $F_{2}=(1000 p)^{1 / 3}$

| $p$ | $F_{2}$ | $p$ | $F_{2}$ | $p$ | $F_{\mathbf{2}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0.0040 | 1.59 | 0.0095 | 2.12 | 0.0150 | 2.47 |
| 0.0045 | 1.65 | 0.0100 | 2.15 | 0.0155 | 2.49 |
| 0.0050 | 1.71 | 0.0105 | 2.19 | 0.0160 | 2.52 |
| 0.0055 | 1.76 | 0.0110 | 2.22 | 0.0165 | 2.54 |
| 0.0060 | 1.82 | 0.0115 | 2.26 | 0.0170 | 2.57 |
| 0.0065 | 1.87 | 0.0120 | 2.29 | 0.0175 | 2.60 |
| 0.0070 | 1.91 | 0.0125 | 2.32 | 0.0180 | 2.62 |
| 0.0075 | 1.96 | 0.0130 | 2.35 | 0.0185 | 2.64 |
| 0.0080 | 2.00 | 0.0135 | 2.38 | 0.0190 | 2.67 |
| 0.0085 | 2.04 | 0.0140 | 2.41 | 0.0195 | 2.69 |
| 0.0090 | 2.08 | 0.0145 | 2.44 | 0.0200 | 2.71 |

Table 4-Values of $F_{3}=\frac{(1000 p)^{(1 / 3+0.4 H / V)}}{(10)^{0.8 H / V}}$

| $H / V$ | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 | 1.0 | 1.1 | 1.2 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| $\boldsymbol{p}$ |  | 1.40 | 1.23 | 1.08 | 0.95 | 0.83 | 0.73 | 0.64 | 0.57 | 0.50 | 0.44 | 0.38 |
| 0.34 |  |  |  |  |  |  |  |  |  |  |  |  |
| 0.0040 | 1.46 | 1.29 | 1.14 | 1.00 | 0.89 | 0.78 | 0.69 | 0.61 | 0.54 | 0.48 | 0.42 | 0.37 |
| 0.0045 | 1.52 | 1.34 | 1.19 | 1.06 | 0.94 | 0.83 | 0.74 | 0.66 | 0.58 | 0.52 | 0.46 | 0.40 |
| 0.0050 | 1.57 | 1.40 | 1.25 | 1.11 | 0.99 | 0.88 | 0.78 | 0.70 | 0.62 | 0.55 | 0.49 | 0.44 |
| 0.0055 | 1.56 |  |  |  |  |  |  |  |  |  |  |  |
| 0.0060 | 1.62 | 1.45 | 1.30 | 1.16 | 1.04 | 0.92 | 0.83 | 0.74 | 0.66 | 0.59 | 0.53 | 0.47 |
| 0.0065 | 1.67 | 1.50 | 1.34 | 1.20 | 1.08 | 0.97 | 0.87 | 0.78 | 0.70 | 0.62 | 0.56 | 0.50 |
| 0.0070 | 1.72 | 1.55 | 1.39 | 1.25 | 1.12 | 1.01 | 0.91 | 0.82 | 0.73 | 0.66 | 0.59 | 0.53 |
| 0.0075 | 1.76 | 1.59 | 1.43 | 1.29 | 1.16 | 1.05 | 0.95 | 0.85 | 0.77 | 0.69 | 0.63 | 0.56 |
| 0.0080 | 1.81 | 1.63 | 1.48 | 1.34 | 1.21 | 1.09 | 0.99 | 0.89 | 0.80 | 0.73 | 0.66 | 0.59 |
| 0.0085 | 1.85 | 1.68 | 1.52 | 1.38 | 1.25 | 1.13 | 1.02 | 0.93 | 0.84 | 0.76 | 0.69 | 0.62 |
| 0.0090 | 1.89 | 1.72 | 1.56 | 1.41 | 1.28 | 1.17 | 1.06 | 0.96 | 0.87 | 0.79 |  | 0.72 |
| 0.65 |  |  |  |  |  |  |  |  |  |  |  |  |
| 0.0095 | 1.93 | 1.75 | 1.60 | 1.45 | 1.32 | 1.20 | 1.10 | 1.00 | 0.91 | 0.83 | 0.75 | 0.68 |
| 0.0100 | 1.96 | 1.79 | 1.63 | 1.49 | 1.36 | 1.24 | 1.13 | 1.03 | 0.94 | 0.86 | 0.78 | 0.71 |
| 0.0105 | 2.00 | 1.83 | 1.67 | 1.53 | 1.40 | 1.27 | 1.16 | 1.06 | 0.97 | 0.89 | 0.81 | 0.74 |
| 0.0110 | 2.04 | 1.86 | 1.71 | 1.56 | 1.43 | 1.31 | 1.20 | 1.10 | 1.00 | 0.92 | 0.84 | 0.77 |
| 0.0115 | 2.07 | 1.90 | 1.74 | 1.60 | 1.46 | 1.34 | 1.23 | 1.13 | 1.04 | 0.96 | 0.87 | 0.80 |
| 0.0120 | 2.10 | 1.93 | 1.78 | 1.63 | 1.50 | 1.38 | 1.26 | 1.16 | 1.07 | 0.98 | 0.90 | 0.83 |
| 0.0125 | 2.14 | 1.96 | 1.81 | 1.66 | 1.53 | 1.41 | 1.30 | 1.19 | 1.10 | 1.01 | 0.93 | 0.86 |
| 0.0130 | 2.17 | 2.00 | 1.84 | 1.70 | 1.56 | 1.44 | 1.33 | 1.22 | 1.13 | 1.04 | 0.96 | 0.88 |


tally to $H / V=0$ and vertically downward to the $p$ scale.
$p=0.98 \%$ OK since it is $<2.0 \%$,
and $>A_{s}+A_{v}=0.4 \%+\frac{0.4 \%}{2}=$ $0.6 \%$.

- Select $A_{s}, A_{v}$, and corbel dimensions.

$$
p=\frac{A_{s}+A_{v}}{b d}=\frac{1.5 A_{s}}{b d},
$$

if $A_{v}$ is made equal to $0.5 A_{s}$.
Hence

$$
\begin{aligned}
& A_{s}=0.0098 \times 14 \times 13.3 / 1.5 \\
&=1.22 \mathrm{in} .^{2} \\
& \text { Use } 4-\# 5 \text { bars } . \\
& A_{v}=0.61 \mathrm{in} .^{2}
\end{aligned}
$$

Use 2-\#4 bar closed stirrups. Stirrup spacing (2-\#4 stirrups)

$$
\begin{aligned}
s= & \frac{2}{3}\left(\frac{d}{n+1}\right)=\frac{2}{3}\left(\frac{13.3}{3}\right) \\
= & 2.96 \mathrm{in} . \\
& \text { Use } 3 \text { in. ctrs. from tension } \\
& \text { reinforcement. }
\end{aligned}
$$

Allowing 1 in. cover to reinforcement, over-all depth of corbel
$=1+0.3+13.3=14.6$ in.
Use 15 in .
Length of corbel

$$
\begin{aligned}
& =2+(\text { bearing width })+(\text { clear }- \\
& \text { ance }) \\
& =2+4+1=7 \mathrm{in} .
\end{aligned}
$$

Depth of outer face of corbel, say half over-all depth at column face,

$$
\begin{aligned}
= & 15 / 2=7.5 \mathrm{in} . \\
& \text { Use } 8 \mathrm{in.} .
\end{aligned}
$$

- Check design.
$d=15.0-1.0-0.62 / 2=13.7 \mathrm{in}$.
$a / d=4.0 / 13.7=0.29$
$p=\frac{A_{s}+A_{v}}{b d}=\frac{2.04}{14 \times 13.7}=1.06 \%$
$V_{u}=\phi b d \sqrt{f_{c}^{\prime}} F_{1} F_{2}$

Using Tables 2 and 3 to obtain $F_{1}$ and $F_{2}$

$$
\begin{aligned}
V_{u}= & 0.85 \times 14 \times 13.7 \times \sqrt{5000} \\
& \times 5.90 \times 2.19
\end{aligned}
$$

$=149 \mathrm{kips}$ OK, greater than required design load.

- The details of this corbel are shown in Fig. 4(a).


## Example 2

Redesign the corbel of Example 1 assuming that a bearing shoe in the prestressed girder is welded to the corbel, and because of this, a horizontal force of 45 kips will occur due to restraint of creep and shrinkage deformation of the girder. This example is illustrated by Fig. 4(b).

- From Example 1, $V_{w}=138$ kips, and $a=4 \mathrm{in}$.
- Section 1506(a)5, of the ACI Building Code (ACI 318-63), requires that the effects of creep and shrinkage be considered on the same basis as the effects of dead load, when calculating the design ultimate loads. Hence the load factor for the horizontal restraint force will be:

$$
\begin{aligned}
& \frac{4}{3} \quad(1.5)=2.0 \\
& H_{u}=2.0(45)=90 \mathrm{kips}
\end{aligned}
$$

therefore

$$
H / V=90 / 138=0.65
$$

- From the design chart, the value of $v_{u}$ corresponding to the maximum allowable $p(=1.3 \%), H / V$ of 0.65 , an assumed $a / d$ of 0.3 , and $f_{c}^{\prime}$ of 5000 psi , is about 460 psi .

Therefore

$$
d=\frac{V_{u}}{v_{u} b}=\frac{138,000}{460 \times 14}=21.4 \mathrm{in}
$$

now

$$
a / d=4 / 21.4=0.19
$$



Fig. 4-Corbel Details for Example Problems

- Determine $p$ from the design chart
for $v_{u}=460 \mathrm{psi}, f_{c}^{\prime}=5000 \mathrm{psi}, a / d=$ 0.19 and $H / V=0.65$.
$p=1.07 \%$ OK since it is $>0.4 \%$ and $<1.3 \%$
- Select $A_{s}, A_{v}$, and corbel dimensions.

$$
p=A_{s} / b d
$$

Hence

$$
\begin{aligned}
A_{*} & =p b d=0.0107 \times 14 \times 21.4 \\
& =3.20 \mathrm{in} .^{2} \\
& \text { Use } 4-\# 8 \text { bars } .
\end{aligned}
$$

$$
\mathrm{A}_{n}=\mathrm{A}_{s} / 2=1.60 \mathrm{in} .{ }^{2}
$$

Use 4-\#4 bar closed stirrups. Stirrup spacing (4-\#4 stirrups)

$$
\begin{aligned}
s & =\frac{2}{3}\left(\frac{d}{n+1}\right)=\frac{2}{3}\left(\frac{21.4}{5}\right) \\
& =2.88 \mathrm{in} . \text { Use } 3 \text {-in. centers. }
\end{aligned}
$$

Assuming a l-in. thick bearing plate welded to the main tension reinforcement, over-all depth of corbel:

$$
\begin{aligned}
& h=1+0.5+21.4=22.9 \mathrm{in} \\
& \text { Use } 23 \mathrm{in.}
\end{aligned}
$$

Length of corbel will be as in Example 1, 7 in.

- Check design.

$$
\begin{align*}
d & =23-1-0.5=21.5 \mathrm{in} \\
a / d & =4 / 21.5=0.19 \\
p & =A_{s} / b d=3.16 /(14 \times 21.5) \\
& =1.05 \% \\
V_{u} & =\phi b d \sqrt{f_{c}^{\prime}} F_{1} F_{3} \tag{2a}
\end{align*}
$$

Using Tables 2 and 4 to obtain $F_{1}$ and $F_{3}$

$$
\begin{aligned}
V_{u}= & 0.85 \times 14 \times 21.5 \times \sqrt{5000} \\
& \times 6.33 \times 1.21
\end{aligned}
$$

$=139$ kips OK, greater than design load

- The details of this corbel are shown in Fig. 4(b).


## Special Note

It should be noted that the addition of the horizontal restraint force has necessitated an increase in depth of the corbel of 53 percent and an increase in main tension reinforcement of 162 percent. It is clear, therefore, that for safety, a realistic estimate must be made of any horizontal forces that may act on a corbel. If special provision is not
made to eliminate the horizontal restraint forces by using lubricated sandwich pads at one end of each girder, it is proposed that $H / V$ should be assumed in design to be at least 0.5 , unless the horizontal force is calculated.

## PART 2-TESTS OF CORBELS

## Scope

Three series of tests were made: (a) exploratory tests, (b) tests of corbels subjected to vertical loads only, and (c) tests of corbels subjected to combined vertical and horizontal loads. The exploratory tests involved testing procedures and reinforcing
detailing. The other two series involved a systematic investigation of the effect of different variables on the strength and behavior of corbels.

The variables considered in the tests were: reinforcement ratio, concrete strength, ratio of shear span to effective depth, amount and distribution of stirrup reinforcement, size and shape of corbel, and the ratio of the horizontal applied load to the vertical applied load. The range of the variables is indicated on Fig. 5.

## Test Specimens

All specimens consisted of a length of $8 \times 12-\mathrm{in}$. column with two corbels


Fig. 5-Corbel Test Specimen
arranged symmetrically, as shown in Fig. 5. With the exception of certain specimens in series (a) the main tension reinforcement consisted of straight deformed bars anchored by bars of equal diameter welded across their ends, as shown in Fig. 6. Corbels with horizontal stirrups were detailed as shown in Fig. 6(b). Corbels to be subjected to combined vertical and horizontal loading were provided with grooved bearing plates welded to the tension reinforcement as shown in Fig. 6(c). The detailing of the reinforcement of the corbels in the exploratory series (a) was as indicated in Fig. 7.

The dimensions of the individual specimens and the material properties are set out in Tables Al through A4 appended to this paper.

## Materials and Fabrication

All concrete was made with Type I portland cement. The coarse aggregate was a gravel of $3 / 4 \mathrm{-in}$. maximum size, and the fine aggregate was Elgin sand. The concrete slumps varied from $l^{1 / 2}$ to 3 in . An air-entraining agent was added to produce 4 to 6 percent air. One batch of concrete was used for each specimen, with the exception of two large specimens, which required two batches each. Three $6 \times 12-\mathrm{in}$. cylinders were taken from each batch for determination of concrete strength. The specimens and test cylinders were moist cured for three days under a plastic cover, and then stored at $70^{\circ} \mathrm{F}$ and 50 percent relative humidity, and were tested at six days. The concrete cylinder strengths


Fig. 6-Reinforcement Details of Test Corbels


Fig. 7-Detailing of Corbel Reinforcement in Auxiliary Test Series
varied from 2110 psi to 6680 psi , as given in Tables Al through A4.

The reinforcing steel conformed to ASTM Designation A305 for deformations. The steel yield strengths were determined from tension tests of $30-\mathrm{in}$. coupons taken from each reinforcing bar used; the yield strength varied from 39,900 psi to $95,800 \mathrm{psi}$, and are given in Tables A1 through A4.

## Instrumentation

The corbels were instrumented with SR4-A-12 strain gages mounted on the reinforcement and with SR4-A-9-4 strain gages mounted on the concrete. This instrumentation varied according to the purpose of individual tests.

## Test Procedures

For convenience all corbels were tested in an upside-down position. A heavily-reinforced U-frame cen-
tered under the loading platen of a million-pound testing machine was used to support the corbels. To assure adequate bearing capacity of the legs of the U-frame, the top of the legs was armored by steel plates. These plates were carefully aligned in the forms of the U-frame before placing the concrete to provide parallel bearing surfaces.

The corbels were subjected to various combinations of vertical and horizontal loads. The loads were increased in increments until failure. After each load increment the development of cracks was observed and marked on the specimens. All strain measurements were recorded continuously by strip-chart strain recorders

## Vertical Loading Only

The corbels were loaded through steel bearing plates placed symmet-


Fig. 8-Vertical Loading of Corbels
rically on the top * of the corbels as shown in Fig. 8. The length of the bearing plates was equal to the width of the corbels. The width $w$ was either 3 or 5 in . and the thickness of the plates was 1 or $1 \frac{1}{2} \mathrm{in}$. To eliminate restraint of deformations, a half-round and a round bar were placed between the bearing plates and another set of steel plates which rested on the supporting U-frame. The load was applied to the bottom of the column stub by the testing machine platen.
To assure uniform load distribu-

[^2]tion on all bearing areas, new plywood inserts were used in each test. A $3 / 4$-in. plywood sheet was placed between the column bottom and the testing machine platen, $1 / 8-\mathrm{in}$. plywood sheets between the corbels and the bearing plates, and $1 / 2$-in. plywood sheets between the U-frame and the second set of steel plates. After the application of the first $10,000 \mathrm{lbs}$, the machine platen was blocked to prevent its rotation.

In the first five tests the load applied to the corbels was checked by load measuring cells to establish that the load was distributed equally to the two corbels. Since the two loads did not differ by more than two percent, the use of these load
measuring cells was discontinued in further tests. The two test setups are shown schematically in Fig. 8. Fig. 9 shows the test setup used for the tests involving vertical load only.


Fig. 9-Test Setup for Vertical Load Only, $H / V=0$

Three tests were made to determine whether a column load carried from upper floors influenced the strength of the corbels. In these tests a load was applied to the top of the column stub by a 100 -ton hydraulic ram. A constant ratio of the machine load to the ram load was maintained throughout each of these tests. The loading of the ram was controlled by the oil pressure indicator but the load was also continuously monitored by a load-measuring cell placed between the ram and the column top. This test setup was similar to that shown in Fig. 9, except for the 100 -ton hydraulic ram which was within the U-portion of the test frame.

## Combined Vertical and Horizontal Loading

The horizontal forces which develop in precast beams as a result of restrained volume changes were simulated by horizontal forces applied at the level of the top of the corbels. To permit a direct transfer of the horizontal forces to the tension reinforcement, the 3 -in. bearing plates were welded to the reinforcing bars. The horizontal forces were applied by four or six hydraulic rams to a set of loading plates, and transferred to the bearing plates through milled shear keys. The hydraulic rams were positioned on each side of the corbels in such a manner that the resultant of the ram loads was at the level of the top of the corbel. The frictional restraint to lateral deformations was eliminated by placing $2-\mathrm{in}$. diameter round bars between the loading plates and the steel plates on the supporting U-frame.
The rams used for applying the horizontal forces were calibrated so that the loads could be correlated with the oil pressure. The operation of the rams during testing was checked by load measuring cells which indicated that the errors in the load as determined from the oil pressure were less than one percent. Therefore, the use of the load-measuring cells was discontinued.
The vertical load was applied in the same manner as in the tests of corbels subjected to vertical loads only. A constant ratio between the vertical and the horizontal loads was maintained throughout each test.

The loading system for combined horizontal and vertical loading is shown in Fig. 10.

## Test Results

The principal data obtained in


Fig. 10-Test Setup for Combined Horizontal and Vertical Loading, $H / V$ does not equal zero
these tests have been listed in Tables Al through A4 appended to this paper. Other data are reproduced where appropriate in the discussion of the behavior of the corbels set out in Part 3.

PART 3-BEHAVIOR OF CORBELS
Series (a)-Exploratory Tests
Effect of Additional Column Loads
Three tests were made on pairs of identical specimens. One of each of the companion specimens was subjected to vertical loads applied to the corbels only, while the other specimen was subjected also to an additional load applied at the top of the column stub. The pertinent data are given in Table 5. These tests show that the strength of the corbels is not significantly influenced by the additional load carried by the
column. Therefore, subsequent tests were performed with loads applied to the corbels only.

## Detailing the Corbel Reinforcement

Test of corbels reinforced conventionally according to Fig. 7(a) have shown the weakness of such detailing when loads were applied close to the outer edges of the corbels. These corbels failed along a surface following the bends of the reinforcement, Fig. 11, indicating that the


Fig. 11-Conventionally Reinforced Corbel, Type BI, Loaded Near Outer Edge. Failure Plane Follows Bend Radius

Table 5-Effects of Additional Column Load

| Specimen | Concrete <br> Strength, $f_{c}^{\prime}$ <br> psi | Effective Depth <br> Shear Span | Load per <br> Corbel at <br> Ultimate, kips | Additional <br> Column <br> Load, kips |
| :---: | :---: | :---: | :---: | :---: |
| 4 | 3520 | 0.171 | 99.9 | 0 |
| 5 E | 4010 | 0.171 | 114.3 | 114.0 |
| 15 | 4500 | 0.370 | 72.0 | 0 |
| 6 E | 4140 | 0.370 | 63.9 | 48.0 |
| 24 | 4250 | 0.372 | 88.8 | 0 |
| 7 E | 4490 | 0.372 | 109.3 | 110.0 |

reinforcement was not fully effective and that it even created a possible source of weakness. Measurements of the strains in the reinforcement along the compression side indicated only small compressive stresses throughout the length of the reinforcement.
Previous tests of corbels and of deep beams ${ }^{8,9,11}$, and the tests reported herein, show that the stresses in the tension reinforcement of a corbel do not vary significantly along its length between the face of the column and the point of load application. Consequently, high bond stresses exist in the outer parts of the tension reinforcement and may lead to bond failures. Such bond failures were observed in tests of deep beams ${ }^{8}$. The anchorage of the bars can be assured by cross-bars welded to the ends of the tension reinforcement as shown in Fig. 7(b). This method proved satisfactory and subsequent tests were made on specimens reinforced with straight tension bars anchored by the welded cross-bars.

Tests of corbels with inclined compression reinforcement welded to the ends of the tension reinforcement, Fig. 7(c), show the compression reinforcement contributes little to the strength of the corbels. Therefore, compression reinforcement was not used in further tests.

The strength of corbels with the three types of reinforcement is compared in Table A1. The specimens designated by letters WC had tension reinforcement with welded cross bars;: Fig. 7(b), specimens BI had bent reinforcement, Fig. 7(a), and specimens WI had compression reinforcement and cross-bars welded to the ends of the tension reinforcement, Fig. 7(c).

The arrangement and amount of
reinforcement in the column has little influence on the strength of the corbels projecting from the column, as may be seen in Table Al. Thus, the amount of column reinforcement used in subsequent tests was that which would prevent failure of the column portion of the test specimens.

## Series (b)-Corbels Subject to Vertical Loads Only

## Behavior Under Load

Initially the corbels behaved elastically, and the stress in the main tension reinforcement was proportional to load. In all the tests, the first cracks to appear were flexural cracks starting at the junction of the horizontal face of the corbel and the face of the column. After formation of these cracks the tension reinforcement stress increased much more rapidly. Typical relationships between applied load and force in the tension reinforcement are shown in Fig. 12. Subsequent development of the cracks depended primarily on the reinforcement ratio and the ratio of the shear span to the effective depth, and was also closely related to the mode of failure.

Four principal types of failure were observed, as described below.

- Flexural Tension-A flexural tension failure occurs by crushing of the concrete at the bottom of the sloping face of the corbel after extensive yielding of the tension reinforcement. Such a failure is illustrated in Fig. 13(a). The appearance of a corbel after a flexural tension failure is characterized by very wide flexural cracks.
- Flexural Compression-A flexural compression failure occurs when crushing of the concrete takes place at the bottom of the corbel before


Fig. 12-Relationship Between Applied Load and Tension Steel Force, Vertical Load Only


Fig. 13-Flexural Failures, $H / V=0$
extensive yielding of the reinforcement has occurred. The tension reinforcement stress at failure is either below or just at the yield point and the flexural cracks, while well developed, have not opened excessively. Such a failure is illustrated in Fig. 13(b).

- Diagonal Splitting-The diagonal splitting mode of failure is shown in Fig. 14(a) and 14(b). The flexural crack pattern was well developed before the diagonal splitting of the concrete, which occurred along a line extending from the bearing plate toward the junction of the sloping face of the corbel and the face of the column. A corbel with such a crack usually fails by shearcompression of the concrete compression zone, as in the corbel shown in Fig. 14(b).
- Shear Failure-Shear failures were characterized by the develop-
ment of a series of short inclined cracks along the plane of the interface between the column and the corbel, as may be seen in Figs. 15(a) and (b). The final failure was by shearing along this weakened plane, and the appearance after failure can be seen in Fig. 15(b).
- Secondary Modes of FailureFailures which did not involve the deepest section of the corbel at the column face were considered secordary modes of failure. These were of two types: (a) the splitting away of a portion of the concrete due to a major crack intersecting the sloping face of the corbel, as seen in Fig. 16(a), and (b) bearing failures of the concrete beneath the bearing plate, as seen in Fig. 16(b). Both types of secondary modes of failure occurred at loads lower than those at which failure would have occurred by one of the principal modes


Fig. 14-Diagonal Splitting Failures (DS), $H / \mathbf{V}=0$

(a)

(b)

Fig. 15-Shear Failures (S), H/V = $\mathbf{0}$

(a) Corbel End Failure (CE)

(b) Bearing Failure (B)

Fig. 16-Secondary Modes of Failure, $\mathbf{H} / \mathbf{V}=\mathbf{0}$
of failure had the secondary failures been prevented.

## Discussion of Behavior

To understand the behavior of corbels and to arrive at design equations, extensive plotting of test data was made. During such studies, further tests were conducted to cover adequate ranges of the significant variables. Empirical design equations were gradually arrived at by numerous comparisons of observed properties to those computed by various expressions. An LGP-30 electronic computer was used in these studies.
The relationships between tension reinforcement force and applied load shown in Fig. 12 are for corbels made from concrete with a strength of about 4000 psi . Similar relationships were found to hold for corbels without stirrups made from 2000 and 6000 -psi concrete. It was not con-
sidered necessary to test corbels with stirrups made from concretes having strengths other than 4000 psi. It was found that the tension reinforcement force, $A_{s} f_{s}$, is a function of the applied load, $V$, of the ratio of shear span to effective depth, $a / d$, and of the concrete strength $f_{c}^{\prime}$. The relationship between load $V$ and tension force $A_{s} f_{s}$ can be idealized as shown in Fig. 17. The linear part of the

$A_{s} f_{s}$ - FORCE IN TENSION REINFORCEMENT
Fig. 17-Idealized Relationship Between Applied Load and Force in Tension Reinforcement
relationship between first cracking and yield of the reinforcement can be represented by the equation:

$$
V=V_{0}+m A_{s} f_{s}
$$

where $V=$ applied load

$$
\begin{aligned}
V_{0} & =\text { nominal cracking load } \\
m & =\text { slope }
\end{aligned}
$$

The nominal cracking load, $V_{o}$, and the slope, $m$, are both functions of $f_{c}^{\prime}$ and $a / d$. These functions can be expressed as:

$$
V_{0}=b d \frac{4.4}{(a / d)^{1 / 2}}\left(\frac{f_{c}^{\prime}}{a / d}\right)^{1 / 3} C_{1}
$$

and $\quad m=\frac{2}{3} \sqrt{\frac{f_{c}^{\prime}}{1000}} \frac{1}{C_{2}}$
Substituting for $V_{0}$ and $m$ in Eq. (3) above yields:

$$
\begin{align*}
v= & \frac{V}{b d}=\frac{4.4}{(a / d)^{1 / 2}}\left(\frac{f_{c}^{\prime}}{a / d}\right)^{1 / 3} C_{1} \\
& +\frac{2}{3} \sqrt{\frac{f_{c}^{\prime}}{1} 000} \frac{p f_{s}}{C_{2}} \tag{4}
\end{align*}
$$

where $C_{1}=1$ for vertical loads only

$$
C_{2}=0.8(10)^{a / 3 d} \text { when there }
$$ are no stirrups

$=0.25(10)^{)^{\alpha / a}}$ when there are stirrups
Eq. (4) may be used to calculate the nominal shear stress, $v$, at working load by substituting the allowable steel stress for $f_{8}$, and can be used to calculate the nominal shear stress at yield of the tension reinforcement, $v_{y}$, by substituting $f_{y}$ for $f_{s}$.

Eq. (4) has been used to calculate the nominal shear stress, $v_{y}$, at yield of tension reinforcement in those corbels tested in which yielding occurred. The average value of ( $v_{y}$ test/ $v_{y}$ calc) given in Table A2 is 1.06 and the standard deviation is 0.135 . When the computed steel stress, $f_{s u}$, given by Eq. (5) below was less than the yield point of the steel used, no value for $v_{y}$ calc is
given in Table A2.
Eq. (4) can also be used to define whether or not the tension reinforcement will yield prior to the corbel developing its ultimate strength. If the nominal shear stress at ultimate strength is $v_{u}\left(=V_{u} / b d\right)$, then transposing Eq. (4) and substituting $v_{u t}$ for $v$ yields:

$$
\begin{align*}
f_{s u}= & {\left[v_{u}-\frac{4.4}{(a / d)^{s_{2}}}\left(\frac{f_{c}^{\prime}}{a / d}\right)^{1 / 3} C_{1}\right] } \\
& \times \frac{1.5 C_{2}}{p \sqrt{f_{c}^{\prime} / 1000}} \tag{5}
\end{align*}
$$

in which the stress, $v_{u}$ may be calculated from Eq. (7).
The tension reinforcement will yield if $f_{s u}$ calculated using Eq. (5) is equal to or greater than the yield point stress $f_{y}$.
To facilitate the use of Eqs. (4) and (5), values of $C_{1}$ and $C_{2}$ have been listed in Tables A7 and A8 appended to this report.

For purposes of practical design, it should usually not be necessary to check the stress in the tension reinforcement. As indicated in the discussion of design criteria in Part 1, yield of the tension reinforcement will usually take place at $7 / 3$ to 1 times the ultimate load. The proposed ultimate strength procedure accounts for this by specifying load factors $1 / 3$ greater than those used for the individual precast members.

## Ultimate Strength

The ultimate strength equation must of necessity be empirical because of the complexity of the state of stress in the corbel. Several conclusions concerning the effect of individual variables on the strength of corbels can be drawn on the basis of the experimental data presented herein. These conclusions, together with the requirements of the laws of
similitude, lead to a suitable form for the ultimate strength equation.

The ultimate strength of a corbel, $V_{u}$, is a function of its width $b$ and effective depth $d$, of the reinforcement ratio, $p\left(=A_{s} / b d\right)$, of the concrete strength $f_{e}^{\prime}$ and of the ratio of the shear span to the effective depth, $a / d$. From the laws of similitude it is concluded that the ultimate strength, $V_{u}$, must be directly proportional to the width $b$ and to the effective depth $d$. The tests have shown that the strength is also proportional to $\sqrt{f_{c}^{\prime}}$. Accordingly, the strength may be expressed in terms of the non-dimensional ratio $V_{u} / b d \sqrt{f_{c}^{\prime}}$. This ratio must be a function of the remaining two variables, $a / d$ and $p$.

The tests show that increasing the $a / d$ ratio lowers the corbel strength, $V_{u}$. The maximum strength is obtained for $a=0$, while $a=\infty$ represents the condition of pure bending. Hence, $V_{u}=0$ when $a=\infty$. The variation of the strength with $a / d$ can be represented by a term of the form $K_{1}\left(1-K_{2}{ }^{d / a}\right)$, where $K_{2}$ is less than unity.

These tests also show that the strength increases when the reinforcement ratio increases. The effect of the reinforcement ratio can be expressed by the term $K_{3} p^{\pi_{4}}$. The foregoing analysis leads to the expression:

$$
\begin{equation*}
\frac{V_{u}}{b d \sqrt{f_{c}^{\prime}}}=K_{1}\left(1-K_{2}^{d / a}\right) K_{3} \mathrm{p}^{K_{4}} \tag{6}
\end{equation*}
$$

The constants $K_{1}$ and $K_{3}$ need not be known separately and may be combined into a single coefficient. Statistical analysis of the test data resulted in the following equation:

$$
\begin{align*}
& \frac{v_{u}}{\sqrt{f_{c}^{\prime}}}=\frac{V_{u}}{b d \sqrt{f_{c}^{\prime}}} \\
& =6.5\left(1-0.5^{d / a}\right)(1000 p)^{1 / 3} \tag{7}
\end{align*}
$$

Multiplying both sides of Eq. (7) by $b d \sqrt{f_{c}^{\prime}}$ and introducing the strength reduction factor $\phi$ yields Eq. (1) of the proposed criteria for design of corbels.

Eq. (7) was used to calculate the nominal shear stress at ultimate strength, $v_{u}$, for all corbels subjected to vertical loads only, and the results of these calculations are listed in Table A2. Excluding those specimens which experienced secondary failures by bearing or splitting off of the corbel end, the average value of ( $v_{u}$ test $/ v_{u}$ calc) was found to be 1.02 , and the standard deviation 0.119.

Analysis of data from tests of corbels with horizontal stirrups shows that the stirrups are as effective in resisting vertical loads as is the main tension reinforcement. Accordingly, the strength of a corbel with horizontal stirrups and subject to vertical loads only can be calculated using Eq. (7) but calculating $p$ on the basis of the total cross section of tension and stirrup reinforcement, i.e. $p=\left(A_{s}+A_{v}\right) / b d$. The calculated ultimate strengths of corbels with stirrups and subject to vertical loads listed in Table A4 were determined in this manner. The average value of ( $v_{u}$ test $/ v_{u}$ calc) was 1.11 and the standard deviation 0.084 .
Fig. 18 shows a graphical representation of Eq. (7), together with the corresponding test values. The test results from corbels which experienced secondary failures are not included in this figure.
In Table A5 comparisons have been made between data obtained by other investigators at the Universities of Illinois and Texas, and the ultimate strengths calculated using Eq. (7). A satisfactory agreement is found.


Fig. 18-Ultimate Strength of Corbels

Series (c)-Corbels Subject to Combined Vertical and Horizontal Loads
Discussion of Behavior
The addition of outward horizontal forces to the vertical loads does not change the essential characteristics of behavior, which can still be represented by the idealized diagram of Fig. 17. However, the functions for $V_{0}$ and $m$ must be modified to account for the lower values of the nominal cracking load $V_{0}$ and of the slope $m$ observed in data from tests of corbels subject to combined loading. Typical relationships be-
tween applied load and tension reinforcement force for corbels subjected to combined loading are shown in Fig. 19.
The function for $v$ derived from the data shown in Fig. 19, and from other similar data not presented here, takes the form:

$$
\begin{align*}
& v=\frac{V}{b d}= \\
& \frac{\frac{4.4}{(a / d)^{3 / 2}}\left(\frac{f^{\prime}}{a / d}\right)^{1 / 3} \mathrm{C}_{1}+\frac{2}{3} \sqrt{\frac{f_{c}^{\prime}}{1000}} \frac{p f_{s}}{\overline{C_{2}}}}{1+\frac{2}{3} \bar{V} \sqrt{\frac{f_{c}^{\prime}}{1000}}}
\end{align*}
$$



Fig. 19-Relationship Between Applied Load and Tension Steel Force, Combined Vertical and Horizontal Loading
where $C_{1}=1.5(a / d)^{2 / 3}$, and $C_{2}=$ $0.7(10)^{a / 2 d}$, whether stirrups are present or not. Eq. (8) reduces to Eq. (4) when $H / V=0$, i.e., for vertical load only. However, it should be noted that coefficients $C_{1}$ and $C_{2}$ must then be as defined earlier for Eq. (4).

Eq. (8) has been used to calculate the nominal shear stress, $v_{y}$, at yield of the tension reinforcement in those corbels tested in which yield of the tension reinforcement occurred. The results are given in Table A3. The average value of ( $v_{y}$ test $/ v_{y}$ calc) was 1.04 for $H / V=1 / 2$, and 0.92 for $H / V=1 / 1$, the standard deviations being 0.088 and 0.084 respectively.

As before, by equating Eq. (8) to the nominal shear stress at ultimate strength, $v_{u}$, and transposing, the reinforcement stress at ultimate strength, $f_{s u}$ can be determined.
$f_{s u}=\left[v_{u}\left\{1+\frac{2}{3} \frac{H}{V} \sqrt{\frac{f_{c}^{\prime}}{1000}}\right\}-\frac{4.4}{(a / d)^{1 / 2}}\right.$
$\left.\binom{f_{j}^{\prime}}{a / d}^{1 / 3} C_{1}\right] \frac{1.5 C_{2}}{p \sqrt{f_{c}^{\prime} / 1000}}$
where $C_{1}$ and $C_{2}$ are as defined for Eq. (8) above, and $v_{u}$ is obtained from Eq. (10) below. Values of $C_{1}$ and $C_{2}$ are also listed in Tables A9 and A10 appended to this report.

For purposes of practical design, yield of the tension reinforcement may again be accounted for by the use of load factors $1 / 2$ greater than those specified for individual members.

## Ultimate Strength

The principles used in the derivation of the ultimate strength equation for corbels subjected to vertical loads only apply also to the derivation of an ultimate strength equation for corbels subject to combined
horizontal and vertical loads. The ultimate strength $V_{u}$ must again be proportional to $b$ and $d$, and it may be assumed that it is also proportional to $\sqrt{\overline{f_{c}^{\prime}}}$. The ratio $V_{u} / b d \sqrt{f_{c}^{\prime}}$ is then a function of $a / d, p$ and $H / V$, which should reduce to Eq. (7) when $H / V=0$, i.e. for vertical loads only. The following equation was established after study of the test data, having in mind the above requirements.

$$
\begin{gather*}
c_{u}=\begin{array}{c}
V_{u} \\
b d
\end{array}=6.5 \sqrt{ } f_{c}^{\top}\left(1-0.5^{d / a}\right) \\
\frac{(1000 p)^{(1 / 3+0.4 H / V)}}{(10)^{0.8 H / V}} \tag{10}
\end{gather*}
$$

Eq. (10) was used to calculate the nominal shear stress at ultimate strength for all corbels subjected to combined vertical and horizontal loads, and the ultimate shear stresses so calculated are set out in Table A3. Eq. (2) of the proposed design criteria is based on Eq. (10). Excluding those specimens which experienced secondary failures (i.e., by bearing or by splitting off of the corbel ends), the average value of ( $v_{u}$ test $/ v_{u}$ calc) was 1.05 for $H / V=$ $1 / 2$, and 1.21 for $H / V=1 / 1$, the standard deviation being 0.132 and 0.216 , respectively.
The appearance of typical corbels after failure under combined loading is shown in Figs. 20 and 21.

A limited number of corbels with stirrups were tested under combined loading, and the results are given in Table A4. It was found that the stirrups did not increase the resistance of a corbel to combined loading by as large a proportion as was the case with a corbel subject to vertical load only. Also, the contribution of the stirrups was more erratic, viz. corbels 13 S and 14 S with $0.34 \%$ and


Fig. 20-Flexural Yielding Failure Followed by Crushing of the Concrete (FT), H/V = 1


Fig. 21-Shear Failure (S), $\mathrm{H} / \mathrm{V}=\mathbf{1 / 2}$
$0.93 \%$ of stirrup steel, respectively, and all else the same, gave ultimate shear stresses of 260 and 273 psi . The effectiveness of the stirrups is also apparently a function of the $a / d$ ratio and of the $H / V$ ratio. A considerable program of tests would be necessary to assess the influence of the various factors which apparently influence the effectiveness of stirrups in a corbel subject to combined loading. For the present it was decided that any contribution from the stirrups should be regarded as reserve strength, and should not be taken into account in design. Stirrups do lead to a more ductile form of failure, and hence it was concluded that a minimum amount of stirrups should always be provided.

## Secondary Failures

The following comments apply to both vertical load only and to combined vertical and horizontal loading.

Corbel End Failure-In certain of the tests the depth of the outer face of the corbel was deliberately varied in order to determine the minimum depth necessary to prevent the occurrence of a secondary failure by splitting away of a portion of the concrete at the tip of the corbel. It was found that this type of failure, as shown by Fig. 22, did not occur in those corbels having a depth below the outer edge of the bearing plate greater than about 0.5 the depth of the corbel at the face of the column.

Bearing Failure-Crushing of the concrete below the bearing plate occurred in some of the tests. The bearing stress, $f_{b u}$, at ultimate strength of the corbels is listed in Table A6. Bearing failures occurred


Fig. 22-Corbel End Failure (CE), $\mathrm{H} / \mathrm{V}=\mathbf{1}$
at stresses as low as $0.34 f^{\prime}{ }_{c}$ when the load was applied near the outer edge of the corbel in a combined loading test. However, if the outer edge of the bearing plate was at least 2 in . from the outer face of the corbel, then bearing failures did not occur at bearing stresses less than $0.5 f_{c}^{\prime} 4$ detailed study of bearing stresses was not made. It is believed that $0.5 f_{c}^{\prime}$ is a suitably conservative value.

## CONCLUDING REMARKS

The experimental evidence presented in this paper indicates that the nominal ultimate shear stress, $v_{u}$, in corbels with a shear span to effective depth ratio less than one may exceed the maximum shear stress allowed by Chapter 17 of the ACI Code (ACI 318-63) for beams with $a / d$ ratio greater than one.

The nominal ultimate shear stress in a corbel is a function of the ratio of the shear span to the effective
depth, of the reinforcement ratio, of the concrete strength, and of the ratio of the horizontal and vertical components of the applied loads.
Horizontal forces acting outward from the column significantly reduce corbel strength, and must be considered in the design of a corbel unless special provisions are made for free movements of the supported beams.
Tension reinforcement and horizontal stirrups are equally effective in increasing the strength of a corbel subject to vertical loads only. However, the effective amount of reinforcement is limited.

Loads carried by a column do not affect the corbel strength, nor does the amount or arrangement of column reinforcement.
The results of this investigation have been used as a basis for the formulation of "Proposed Criteria for the Design of Corbels" which is presented in Part 1 of this paper.

## ACKNOWLEDGMENTS

The work described herein was carried out in the Structural Laboratory of the Portland Cement Association under the direction of Eivind Hognestad and Alan H. Mattock. Contributions were made by several members of the laboratory staff. Particular credit is due Bernard J. Doepp, William Hummerich, Jr., David C. Yates and Kenneth Hirte for the laboratory work involved.

## NOTATION

The notation of the ACI Building Code (ACI 318-63) is used wherever applicable. The letter symbols used in this paper are defined below:
$A_{s}=$ area of tension reinforcement, in. ${ }^{2}$
$A_{v}=$ area of horizontal stirrups, in. ${ }^{2}$
$a \quad=$ shear span measured from the face of the column to the resultant of applied load, in.
$b \quad=$ width of corbel, in.
$d \quad=$ effective depth of the centroid of tension reinforcement at the column face, in.
$f_{b u} \quad=$ bearing stress at ultimate strength, psi
$f_{s} \quad=$ stress in tension reinforcement, psi
$f_{s u}=$ stress in tension reinforcement at ultimate strength, psi
$f_{v} \quad=$ stress in horizontal stirrups, psi
$f_{y} \quad=$ yield stress of reinforcement, psi
$f_{c}^{\prime}=$ concrete cylinder strength, psi
$\sqrt{f_{c}^{\prime}}=$ relationship expressed in psi , so that $\sqrt{f_{c}^{\prime}}=60 \mathrm{psi}$ for $f_{c}^{\prime}=3600 \mathrm{psi}$
$H / V=$ ratio ${ }^{c}$ of horizontal and vertical components of applied loads
$h \quad=$ over-all depth of corbel at column face, in.
$h^{\prime} \quad=$ depth of corbel outer face, in.
$n \quad=$ number of horizontal closed stirrups
$p \quad=$ reinforcement ratio $=\left(A_{s}\right.$ $\left.+A_{v}\right) / b d$ when $H / V=0$ $=A_{s} / b d$ when $H / V$ does not equal zero.
$s \quad=$ center to center spacing of stirrups, in.
$V \quad=$ applied vertical load, lb
$v \quad=$ nominal shearing stress $=$ $\frac{V}{b d}$, psi
$V_{0}=$ nominal cracking load, lb
$V_{u}=$ ultimate vertical load, lb
$v_{u} \quad=$ nominal ultimate shearing stress $=V_{u} / b d, \mathrm{psi}$
$V_{y} \quad=$ vertical load at initial yielding of tension reinforcement, lb
$v_{y} \quad=$ nominal shearing stress at initial yielding of tension reinforcement $=V_{y} / b d$, psi
$w \quad=$ width of bearing plates, in.
$\phi \quad=$ capacity reduction factor

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| No. | Type | $\begin{aligned} & \text { h, } \\ & \text { in. } \end{aligned}$ |  | $\begin{aligned} & \text { a, } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & d, \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \mathrm{p} \\ & \% \end{aligned}$ | a/d | $\begin{aligned} & f_{c}^{\prime} \\ & \mathrm{psi} \end{aligned}$ | $\begin{aligned} & \mathrm{f}_{\mathrm{y}}, \\ & \mathrm{ksi} \end{aligned}$ | $\begin{gathered} \mathbf{f}_{\text {su }} \text { te } \\ \mathrm{ksi} \end{gathered}$ | $\begin{aligned} & \mathrm{v}_{\mathrm{y}}^{\mathrm{te}} \\ & \mathrm{psi} \end{aligned}$ | $\begin{gathered} \mathrm{v}_{\mathrm{y}} \mathrm{ca} \\ \mathrm{psi} \end{gathered}$ | $\mathrm{v}_{\mathrm{u}}$ te <br> psi | $\begin{gathered} \mathrm{v}_{\mathrm{u}} \mathrm{c} \\ \mathrm{ps} \end{gathered}$ | $\frac{v_{y} \text { test }}{v_{y} \text { calc }}$ | $\frac{v_{u} \text { test }}{v_{u} \text { calc }}$ | $\frac{e s t / 7}{s t)_{W C}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Effect of Reinforcing Details |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 42年 | WC | 26 | 11 | 9.5 | 24.1 | 0.62 | 0.249 | 4850 | 52.5 | * | * | 605 | 778 | 689 |  | 2.13 | 1.00 |
| 1EI | BI | 26 | 11 | 9.5 | 24.1 | 0.62 | 0.249 | 4190 | 48.5 | 34.0 | - | 539 | 429 | 640 | - | 1.13 0.67 | 1.00 0.59 |
| $2 E^{\frac{1}{1}}$ | WI | 26 | 11 | 9.5 | 24.1 | 0.62 | 0.249 | 4440 | 44.9 | 44.9 | 726 | 520 | 751 | 659 | 1.40 | 1.14 | 1.01 |
| 29 | WC | 26 | 26 | 6.0 | 24.1 | 0.62 | 0.249 | 3730 | 47.5 | 45.3 | - | 611 | 640 | 684 | - | 0.94 | 1.00 |
| $3 \mathrm{E}^{1}$ | BI | 26 | 12 | 6.0 | 24.1 | 0.62 | 0.249 | 3980 | 43.0 | 43.0 | 648 | 480 | 648 | 624 | 1.35 | 1.04 | 0.98 |
| $4 \mathrm{E}^{\frac{1}{1}}$ | WI | 26 | 12 | 6.0 | 24.1 | 0.62 | 0.249 | 4200 | 44.9 | 44.9 | 726 | 507 | 755 | 641 | 1.43 | 1.18 | 1.11 |
| Effect of Additional Column Load |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 4 | WC | 18 | 9 | 2.75 | 16.1 | 0.93 | 0.171 | 3520 | 43.6 | 43.6 | 777 | - |  |  | - |  |  |
| $5 \mathrm{E}^{\mathbf{1}}$, 3 | CL 100\% | 18 | 9 | 2.75 | 16.1 | 0.93 | 0.171 | 4010 | 44.5 | * | * | - | 889 | 851 | - | 1.04 | $\begin{aligned} & 1.00 \\ & 1.07 \end{aligned}$ |
| 15 | WC | 18 | 6 | 6 | 16.2 | 0.48 | 0.370 | 4500 | 48.1 | 48.1 | 405 | 472 | 556 | 622 | 0.86 | 0.89 | 1.00 |
| 6 E | CL 75\% | 18 | 6 | 6 | 16.2 | 0.48 | 0.370 | 4140 | 48.1 | 48.1 | 478 | 456 | 493 | 597 | 1.05 | 0.82 | 0.92 |
| 24 | WC | 18 |  | $6$ | 16.1 | 0.93 | 0.372 | 4250 | 47.3 | 42.5 | - |  |  |  |  |  | 1.00 |
| $7 \mathrm{E}^{\mathbf{1}}$ | CL $100 \%$ | 18 | 9 | 6 | 16.1 | 0.93 | 0.372 | 4490 | 44.5 | 44.5 | 770 | 716 | 850 | 774 | 1.08 | $\begin{aligned} & 0.92 \\ & 1.10 \end{aligned}$ | 1.00 1.20 |
| Effect of Column Reinforcement |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| $8 \mathrm{El}^{1}$ | WC | 26 | 16 | 9.5 | 24.1 | 0.62 | 0.394 | 4580 | 46.5 | * | * | 540 | 726 | 669 | - | 1.08 | 1.00 |
| $9 \mathrm{E}^{2}$ | CR 3-\#\# | 26 | 16 | 9.5 | 24.1 | 0.62 | 0.394 | 4790 | 53.3 | * | * | 608 | 772 | 685 | - | 1.13 | 1.17 |
| $10 \mathrm{E}^{\frac{1}{3}}$ | CR 6-\#9 | 26 | 16 | 9.5 | 24.1 | 0.62 | 0.394 | 4750 | 46.5 | * | * | 549 | 659 | 681 | - | 0.97 | 1.12 |

## NOTES :

K $=12$ in. and $b=8$ in. for all specimens
WC $=$ welded cross-bar tension reinforcement
$\mathrm{BI}=$ inclined bar formed by bending tension reinforcement
WT = inclined bar welded to WC tension reinforcement
CL- \% additional column load, \% indicates ratio of column laad to corbel load
$C R=$ No. following $C R$ indicates reinforcing bars in $8 \times 12^{\prime \prime}$ column
$\underline{1} w=5$ in. (in all other cases $w=3 \mathrm{in}$.)
2 column failed

* not measured or inconclusive test data

| No. | $\begin{aligned} & \text { h, } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \text { h' } \\ & \text { in. } \end{aligned}$ | $\begin{gathered} \text { a, } \\ \text { in. } \end{gathered}$ | $\begin{array}{r} d, \\ \text { in. } \end{array}$ | $\begin{aligned} & \mathrm{p}, \\ & \% \end{aligned}$ | $\frac{a}{a}$ | $\begin{aligned} & f_{c}^{\prime} \\ & p s i \end{aligned}$ | $\mathrm{I}_{\mathrm{y}}$ ksi | $\mathrm{f}_{\text {su }}$ test <br> ksi | $V_{y}$ test, psi | $\begin{gathered} \mathrm{v}_{\mathrm{y}}^{\mathrm{cal}} \\ \mathrm{psi} \\ \hline \end{gathered}$ | $\begin{gathered} v_{u} \text { tes } \\ \text { psi } \end{gathered}$ | $\begin{gathered} \mathrm{v}_{\mathbf{u}} \mathrm{calc}, \\ \mathrm{psi} \\ \hline \end{gathered}$ | $\frac{v_{y^{\text {test }}}}{v_{y^{c a l c}}}$ | $\frac{v_{u} \text { test }}{v_{u} \text { calc }}$ | $\begin{gathered} \text { Type } \\ \text { Failures } \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 22 | 6 | 2.75 | 20.2 | 0.38 | 0.136 | 3790 | 45.3 | 45.3 | 619 | 616 | 619 | 623 | 1.00 | 0.99 | 5 |
| $2 \pm$ | 22 | 6 | 2.75 | 20.2 | 0.38 | 0.136 | 6170 | 47.0 | 47.0 | 990 | 761 | 1090 | 792 | 1.30 | 1.38 | S |
| 3 | 26 | 6 | 2.75 | 24.2 | 0.32 | 0.114 | 3820 | 45.3 | 45.3 | 529 | 4 | 563 | 591 | - | 0.95 | S |
| 4 | 18 | 6 | 2.75 | 16.1 | 0.93 | 0.171 | 3520 | 43.6 | 43.6 | 777 | 4 | 777 | 797 | - | 0.97 | 5 |
| 5 | 22 | 6 | 2.75 | 20.1 | 0.75 | 0.137 | 3840 | 43.3 | 43.3 | 808 | 4 | 849 | 782 | - | 1.08 | S |
| 6 | 26 | 6 | 2.75 | 24.1 | 0.62 | 0.114 | 3970 | 47.0 | 47.0 | 687 | 4 | 713 | 751 | - | 0.95 | 5 |
| 7 | 18 | 6 | 2.75 | 16.1 | 1.86 | 0.171 | 3260 | 43.3 | 29.4 | - | 4 | 1090 | 967 | - | 1.13 | 5 |
| 8 | 22 | 6 | 2.75 | 20.1 | 1.49 | 0.138 | 4170 | 45.8 | 45.8 | 1060 | 4 | 1090 | 1030 | - | 1.06 | 5 |
| $9^{1}$ | 22 | 6 | 2.75 | 20.1 | 1.49 | 0.138 | 6500 | 45.0 | 45.0 | 1650 | 4 | 1650 | 1280 | - | 1.29 | S |
| 10 | 26 | 6 | 2.75 | 24.1 | 1.24 | 0.114 | 4790 | 47.0 | 47.0 | 898 | 4 | 898 | 1040 | - | 0.86 | 5 |
| $11^{2}$ | 14 | 8 | 4.0 | 12.1 | 1.24 | 0.330 | 3900 | 47.7 | 47.7 | 940 | 4 | 954 | 826 | - | 1.15 | DS |
| 12 | 18 | 6 | 6.0 | 16.2 | 0.31 | 0.370 | 4240 | 51.0 | 51.0 | 385 | 366 | 544 | 521 | 2.05 | 1.04 | FT |
| 13 | 22 | 6 | 6.0 | 20.2 | 0.25 | 0.298 | 4580 | 51.0 | 51.0 | 371 | 380 | 594 | 536 | 0.98 | 1.11 | FT |
| 14 | 26 | 6 | 6.0 | 24.2 | 0.21 | 0.248 | 4540 | 51.0 | 51.0 | 310 | 388 | 434 | 524 | 0.80 | 0.83 | DS |
| 15 | 18 | 6 | 6.0 | 16.2 | 0.48 | 0.370 | 4500 | 48.1 | 48.1 | 405 | 472 | 556 | 622 | 0.86 | 0.89 | DS |
| 161 | 18 | 6 | 6.0 | 16.2 | 0.48 | 0.370 | 3430 | 48.0 | 48.0 | 463 | 419 | 606 | 543 | 1.10 | 1.12 | DS |
| 17 | 18 | 6 | 6.0 | 16.2 | 0.48 | 0.370 | 3990 | 95.8 | 72.2 | - | 4 | 660 | 585 | - | 1.13 | DS |
| 18 | 18 | 18 | 6.0 | 16.2 | 0.48 | 0.370 | 4210 | 47.3 | 47.3 | 556 | 454 | 625 | 601 | 1.22 | 1.04 | S |
| 19 | 22 | 6 | 6.0 | 20.2 | 0.38 | 0.297 | 3790 | 43.2 | 43.2 | 526 | 403 | 572 | 565 | 1.30 | 2.01 | DS |
| 20 | 22 | 6 | 6.0 | 20.2 | 0.38 | 0.297 | 3550 | 95.8 | 67.8 | - | 4 | 533 | 547 | - | 0.97 | DS |
| 21 | 26 | 6 | 6.0 | 24.2 | 0.32 | 0.248 | 3920 | 43.2 | 43.2 | 426 | 411 | 491 | 563 | 1.04 | 0.87 | DS |
| 22 | 26 | 6 | 6.0 | 24.2 | 0.32 | 0.248 | 3740 | 95.8 | 66.2 | - | $\underline{4}$ | 542 | 549 | - | 0.99 | DS |
| 23 | 26 | 26 | 6.0 | 24.2 | 0.32 | 0.248 | 3950 | 45.0 | 43.9 | - | 420 | 457 | 566 | - | 0.81 | DS |
| 24 | 18 | 6 | 6.0 | 16.1 | 0.93 | 0.372 | 4250 | 47.3 | 42.5 | - | 731 | 691 | 753 | - | 0.92 | DS |
| 25 | 18 | 6 | 6.0 | 16.1 | 0.93 | 0.372 | 6410 | 46.6 | 46.6 | 893 | 885 | 1010 | 925 | 1.02 | 1.09 | DS |
| 26 | 18 | 18 | 6.0 | 16.1 | 0.93 | 0.372 | 4280 | 53.3 | 53.3 | 859 | $\stackrel{4}{4}$ | 859 | 755 | - | 1.14 | 5 |
| 27 | 22 | 6 | 6.0 | 20.1 | 0.75 | 0.298 | 4320 | 47.3 | 47.3 | 653 | 683 | 715 | 753. | 0.96 | 0.95 | DS |
| 28 | 26 | 6 | 6.0 | 24.1 | 0.62 | 0.249 | 4630 | 47.3 | 47.3 | 648 | 670 | 648 | 762 | 0.97 | 0.85 | DS |
| 29 | 26 | 26 | 6.0 | 24.1 | 0.62 | 0.249 | 3730 | 47.5 | 45.3 | - | 611 | 640 | 684 | - | 0.94 | S |
| 30 | 22 | 6 | 6.0 | 20.0 | 0.99 | 0.300 | 4260 | 45.6 | 45.6 | 844 | 810 | 844 | 820 | 1.04 | 1.03 | DS |
| 31. | 26 | 6 | 6.0 | 24.0 | 0.82 | 0.250 | 4040 | 46.6 | 46.6 | 781 | 753 | 782 | 782 | 1.04 | 1.00 | DS |
| 32 | 26 | 6 | 6.0 | 24.0 | 0.82 | 0.250 | 4390 | 45.6 | 45.6 | 716 | 770 | 729 | 814 | 0.93 | 0.89 | DS |
| 33 | 18 | 6 | 6.0 | 16.1 | 1.86 | 0.372 | 3830 | 47.3 | 27.2 | - | 4 | 885 | 900 | - | 0.98 | DS |
| 34 | 18 | 18 | 6.0 | 16.1 | 1.86 | 0.372 | 4070 | 53.3 | 34.0 | - | 4 | 959 | 928 | - | 1.03 | DS |
| 35 | 22 | 6 | 6.0 | 20.1 | 1.49 | 0.298 | 3820 | 47.3 | * | * | 生 | 822 | 892 | - | (0.92) | B |
| 36 | 26 | 6 | 6.0 | 24.1 | 1.24 | 0.249 | 3960 | 47.3 | * | * | 4 | 804 | 889 | - | 0.90 | DS |
| 37 | 26 | 26 | 6.0 | 24.1 | 1.24 | 0.249 | 3770 | 54.3 | 28.3 | - | 4 | 809 | 867 | - | 0.93 | S |
| 381 | 18 | 6 | 9.5 | 16.1 | 0.93 | 0.590 | 4700 | 53.0 | * | * | 4 | 664 | 647 | - | 1.03 | DS |
| 397 | 18 | 9 | 9.5 | 16.1 | 0.93 | 0.590 | 4490 | 54.5 | * | * | 4 | 674 | 632 | - | 1.07 | DS |
| 401 | 18 | 12 | 9.5 | 16.1 | 0.93 | 0.590 | 4340 | 44.3 | 44.3 | 660 | 567 | 675 | 622 | 1.16 | 1.08 | FC |

Table A2-Test Results for Vertical Load Series (continued)

| No. | $\begin{aligned} & \mathrm{h}, \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \text { h', } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \text { a, } \\ & \text { in. } \end{aligned}$ | d, <br> in. | $\begin{aligned} & \mathrm{p}, \\ & \% \\ & \hline \end{aligned}$ | $\frac{\mathrm{a}}{\text { a }}$ | $\begin{aligned} & f_{c}^{\prime} \\ & \text { psi } \end{aligned}$ | $\begin{aligned} & \mathrm{f}_{\mathrm{y}^{\prime}} \\ & \mathrm{ksi} \end{aligned}$ | $\begin{gathered} \mathrm{f}_{\mathrm{suq}} \mathrm{te} \\ \mathrm{ksi} \end{gathered}$ | $\mathrm{v}_{\mathrm{y}} \mathrm{test},$ psi | $\begin{aligned} & \mathrm{v}_{\mathrm{y}}^{\mathrm{cal}} \\ & \mathrm{psi} \end{aligned}$ |  | $\begin{gathered} \mathrm{v}_{\mathrm{u}} \mathrm{calc}, \\ \mathrm{psi} \\ \hline \end{gathered}$ | $\frac{\mathrm{v}_{\mathrm{y}}^{\mathrm{te}}}{\mathrm{v}_{\mathrm{y}} \mathrm{ca}}$ | $\frac{v_{u} \text { test }}{v_{u} \operatorname{calc}}$ | Type Failures |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $41 \frac{1}{1}$ | 18 | 18 | 9.5 | 16.1 | 0.93 | 0.590 | 4200 | 44.4 | 44.4 | 606 | 560 | 606 | 612 | 1.08 | 0.99 | DS |
| $42 \frac{1}{1}$ | 26 | 11 | 9.5 | 24.1 | 0.62 | 0.394 | 4850 | 52.5 | * | * | 605 | 778 | 689 | - | 1.13 | S |
| 431 | 26 | 16 | 9.5 | 24.1 | 0.62 | 0.394 | 4140 | 45.7 | 45.7 | 622 | 510 | 622 | 636 | 1.22 | 0.98 | 5 |
| 441 | 26 | 26 | 9.5 | 24.1 | 0.62 | 0.394 | 3840 | 45.4 | 45.4 | 581 | 491 | 581 | 613 | 1.18 | 0.95 | DS |
| 451 | 18 | 9 | 9.5 | 16.1 | 1.86 | 0.590 | 4280 | 50.5 | * | * | 4 | 932 | 780 | - | 1.19 | DS |
| 461 | 18 | 12 | 9.5 | 16.1 | 1.86 | 0.590 | 3840 | 44.3 | 32.5 | - | 4 | 814 | 738 | - | 1.10 | DS |
| $47 \frac{1}{1}$ | 18 | 18 | 9.5 | 16.1 | 1.86 | 0.590 | 4060 | 44.4 | 36.8 | - | 4 | 811 | 759 | - | 1.07 | DS |
| $48 \frac{1}{1}$ | 26 | 6 | 9.5 | 24.1 | 1.24 | 0.394 | 4920 | 45.4 | * | * | 4 | 718 | 875 | - | (0.82) | ${ }_{\text {CE }}$ |
| 491 | 26 | 21 | 9.5 | 24.1 | 1.24 | 0.394 | 4180 | 48.0 | * | * | 4 | 716 | 807 | - | (0.89) | B |
| $50^{1}$ | 26 | 26 | 9.5 | 24.1 | 1.24 | 0.394 | 4390 | 45.4 | 21.3 | - | 4 | 477 | 826 | - | (0.58) | B |
| 51.1 | 26 | 26 | 9.5 | 24.1 | 1.24 | 0.394 | 4490 | 45.7 | 22.7 | - | 4 | 563 | 835 | - | (0.67) | B |
| 52 | 18 | 6 | 10.0 | 16.2 | 0.48 | 0.617 | 3960 | 44.3 | 44.3 | 336 | 323 | 370 | 465 | 1.04 | (0.79) | B |
| 53 54 | 18 | 6 | 10.0 | 16.2 | 0.48 | 0.617 | 6360 | 44.3 | 44.3 | 347 | 399 | 420 | 588 | 0.87 | (0.71) | B |
| 54 55 | 18 | 18 | 10.0 | 16.2 | 0.48 | 0.617 | 3950 | 45.0 | 45.0 | 347 | 326 | 347 | 464 | 1.06 | (0.75) | B |
| 55 | 22 | 6 | 10.0 | 20.2 | 0.38 | 0.495 | 4010 | 45.3 | 45.3 | 355 | 324 | 374 | 485 | 1.10 | (0.77) | CE |
| 56 57 | 26 26 | 6 26 | 10.0 10.0 | 24.2 | 0.32 | 0.413 | 3770 | 45.3 | 45.3 | 301 | 314 | 301 | 478 | 0.96 | (0.63) | CE |
| 57 58 | 18 | 26 | 10.0 | 24.2 | 0.32 | 0.413 | 4130 | 47.5 | 47.5 | 337 | 335 | 435 | 499 | 1.00 | (0.87) | B. |
| 59 | 18 | 6 | 10.0 | 16.1 | 0.93 | 0.621 | 3510 | 43.3 | 34.0 | - | 490 | 424 | 545 | - | (0.78) | CE |
| 60 | 18 | 12 | 10.0 | 16.1 | 0.93 | 0.621 | 3820 | 44.3 | 43.9 | - | 520 | 620 | 568 | - | 1.09 | FC |
| 61 | 18 | 18 | 10.0 | 16.1 | 0.93 | 0.621 | 4110 | 54.3 | 46.7 | - | 4 | 583 | 589 | - | 0.99 | DS |
| 62 | 22 | 6 | 10.0 | 20.1 | 0.75 | 0.497 | 3260 | 43.6 | 36.1 | - | 451 | 436 | 545 | - | (0.80) | CE |
| 63 | 26 | 6 | 10.0 | 24.1 | 0.62 | 0.415 | 3420 | 43.6 | 31.8 | - | 442 | 339 | 567 | - | (0.60) | CE |
| 64 | 26 | ${ }^{6}$ | 10.0 | 24.1 | 0.62 | 0.415 | 6540 | 46.6 | 41.0 | - | 621 | 493 | 785 | - | (0.63) | CE |
| 65 | 26 | 16 | 10.0 | 24.1 | 0.62 | 0.415 | 3660 | 53.2 | 39.6 | - | 525 | 519 | 587 | - | (0.88) | B |
| 66 67 | 26 | 26 | 10.0 | 24.1 | 0.62 | 0.415 | 4040 | 44.1 | 38.2 | - | 480 | 481 | 617 | - | (0.78) | B |
| 67 68 | 26 18 | 26 6 | 10.0 10.0 | 24.1 | 0.62 1.86 | 0.415 0.621 | 4060 3380 | 52.8 43.0 | 34.7 23.4 | - | $\underset{4}{54}$ | 458 | 618 | - | (0.74) | B |
| 69 | 18 | 12 | 10.0 | 16.1 | 1.86 | 0.621 | 3680 | 43.0 44.3 | 23.4 29.7 | - | 4 | 418 | 673 702 | - | $(0.62)$ 0.95 | CE |
| 70 | 18 | 18 | 10.0 | 16.1 | 1.86 | 0.621 | 4010 | 53.3 | 25.2 | - | 4 | 621 | 733 | - | 0.85 | S |
| 71 | 22 | 6 | 10.0 | 20.1 | 1.49 | 0.497 | 4410 | 45.6 | 27.6 | - | 4 | 491 | 799 | - | (0.61) | CE |
| 72 73 | 26 | ${ }^{6} 6$ | 10.0 | 24.1 | 1.24 | 0.415 | 4110 | 45.6 | 26.8 | - | 4 | 392 | 784 | - | (0.50) | CE |
| 73 74 | 26 | 16 26 | 10.0 10.0 | 24.1 | 1.24 | 0.415 0.415 | 4050 4360 | 44.1 | 24.1 | - | 4 | 570 | 778 | - | (0.73) | B |
| 74 753 | 26 45 | 26 12 | 10.0 12.5 | 24.1 | 1.24 0.95 | 0.415 0.300 | 4360 4110 | 52.7 45.4 | 26.3 37.4 | - | $\stackrel{4}{4}$ | 599 | 807 | - | (0.74) | B |
| 763 | 45 | 12 | 12.5 | 41.7 | 0.95 | 0.300 0.300 | 4090 | 45.4 46.7 | 37.4 35.3 | - | 770 785 | 641 | 794 | - | 0.81 | DS |
| 771 | 26 | 6 | 3.5 | 24.2 | 0.48 | 0.144 | 2210 | 45.3 | 34.6 | - | $\stackrel{4}{4}$ | 474 | 511 | - | 0.93 | ${ }_{S}$ |
| 781 | 26 | 6 | 3.5 | 24.1 | 0.93 | 0.145 | 2200 | 44.3 | 39.6 | - | 4 | 546 | 636 | - | 0.86 | S |
| $79{ }^{1}$ | 26 | 26 | 3.5 | 24.1 | 1.24 | 0.145 | 2400 | 47.3 | 34.0 | ${ }^{-}$ | 4 | 517 | 732 | - | (0.71) | B |
| $80^{\underline{1}}$ | 22 | 6 | 6.0 | 20.2 | 0.49 | 0.297 | 2430 | 43.5 | 43.5 | 495 | 385 | 515 | 493 | 1.28 | 1.04 | FC |

Table A2-Test Results for Vertical Load Series (concluded)

| No. |  |  | $\begin{aligned} & \text { a, } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \text { d, } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \mathrm{p} \\ & \% \end{aligned}$ | $\frac{a}{d}$ | $\begin{aligned} & \mathbf{n}^{\prime} \\ & \mathrm{psi} \end{aligned}$ | $\begin{aligned} & \mathrm{f}_{\mathrm{y}}, \\ & \mathrm{ksi} \end{aligned}$ | $\mathrm{f}_{\mathrm{su}} \mathrm{ksi}_{\mathrm{ksi}}$ | $\begin{aligned} & v_{y} \text { test, } \\ & \text { psi } \end{aligned}$ | $\begin{aligned} & \mathrm{v}_{\mathrm{y}}^{\mathrm{caj}} \\ & \mathrm{psi} \\ & \hline \end{aligned}$ | $\begin{gathered} v_{u} \text { te } \\ \text { psi } \end{gathered}$ | $\begin{aligned} & \mathrm{v}_{\mathrm{u}} \mathrm{calc}, \\ & \mathrm{psi} \\ & \hline \end{aligned}$ | $\frac{\mathrm{v}_{\mathrm{y}} \text { test }}{\mathrm{v}_{\mathrm{y}} \mathrm{calc}}$ | $\frac{v_{u} \text { test }}{v_{u} \text { calc }}$ | Type <br> Failures |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 811 | 22 | 6 | 6.0 | 20.1 | 0.94 | 0.298 | 2570 | 44.6 | 44.6 | 672 | 610 | 672 | 627 | 1.10 | 1.07 | FC |
| 821 | 18 | 6 | 4.75 | 16.0 | 1.23 | 0.297 | 2110 | 45.1 | 36.3 | - | 4 | 659 | 623 | - | 1.06 | FC |
| 831 | 18 | 18 | 8.5 | 16.2 | 0.48 | 0.525 | 2310 | 45.8 | 45.8 | 386 | 285 | 397 | 386 | 1.35 | 1.03 | FC |
| 341 | 18 | 18 | 8.5 | 16.1 | 0.93 | 0.528 | 2290 | 47.3 | 42.5 | - | 470 | 543 | 478 | - | 1.14 | FC |
| 851 | 18 | 18 | 8.5 | 16.0 | 1.23 | 0.531 | 2170 | 44.6 | 31.1 | - | 4 | 495 | 510 | - | 0.97 | FC |
| 861 | 26 | 6 | 3.5 | 24.2 | 0.48 | 0.244 | 4180 | 46.3 | 46.3 | 826 | 694 | 878 | 703 | 1.19 | 1.25 | 3 |
| $87 \underline{1}$ | 26 | 6 | 3.5 | 24.1 | 0.93 | 0.145 | 3880 | 44.3 | 38.1 | - | 4 | 804 | 845 | - | 0.95 | S |
| $88 \frac{1}{2}$ | 26 | 26 | 3.5 | 24.1 | 1.24 | 0.145 | 3820 | 47.5 | 34.0 | - | 4 | 774 | 923 | - | 0.84 | $s$ |
| $89{ }^{1}$ | 22 | 6 | 6.0 | 20.2 | 0.49 | 0.297 | 4010 | 44.8 | 44.8 | 557 | 487 | 681 | 634 | 1.14 | 1.07 | FC |
| $90^{1}$ | 22 | 6 | 6.0 | 20.1 | 0.94 | 0.298 | 4240 | 46.5 | 46.5 | 964 | 791 | 967 | 805 | 2.22 | 1.20 | 3 |
| 911 | 18 | 6 | 4.75 | 16.0 | 1.23 | 0.297 | 4060 | 46.7 | 45.3 | - | 4 | 961 | 864 | - | 1.11 | 5 |
| 923 | 18 | 18 | 8.5 | 16.2 | 0.48 | 0.525 | 4160 | 45.8 | 45.8 | 347 | 370 | 497 | 518 | 0.94 | 0.96 | DS |
| 931 | 18 | 18 | 8.5 | 16.1 | 0.93 | 0.528 | 3980 | 47.5 | 47.5 | 699 | 610 | 699 | 631 | 1.14 | 1.11 | DS |
| 941 | 18 | 18 | 8.5 | 16.0 | 1.23 | 0.531 | 3940 | 46.7 | 42.5 | - | 4 | 888 | 687 | - | 1.29 | DS |
| 951 | 26 | 6 | 3.5 | 24.2 | 0.48 | 0.144 | 6310 | 45.3 | 45.3 | 826 | 815 | 981 | 864 | 1.01 | 1.14 | 5 |
| 961 | 26 | 6 | 3.5 | 24.1 | 0.93 | 0.245 | 6430 | 46.5 | 46.5 | 1300 | 4 | 1300 | 1090 | - | 1.19 | 5 |
| 97ㄹ | 26 | 26 | 3.5 | 24.1 | 1.24 | 0.145 | 6420 | 44.3 | 32.5 | - | 4 | 1110 | 1200 | - | 0.93 | 5 |
| 981 | 22 | 6 | 6.0 | 20.2 | 0.49 | 0.297 | 6610 | 44.5 | 44.5 | 619 | 603 | 787 | 813 | 1.03 | 0.97 | 5 |
| 991 | 22 | 6 | 6.0 | 20.1 | 0.94 | 0.298 | 6570 | 46.5 | 46.5 | 902 | 967 | 1150 | 1000 | 0.93 | 1.14 | DS |
| 1001 | 18 | 6 | 4.75 | 16.0 | 1.23 | 0.297 | 6430 | 47.5 | 47.5 | 1340 | 4 | 1340 | 1090 | - | 1.23 | 5 |
| 1011 | 18 | 18 | 8.5 | 16.2 | 0.48 | 0.525 | 6370 | 46.3 | 46.3 | 386 | 451 | 602 | 641 | 0.85 | 0.94 | FC |
| 1023 | 18 | 18 | 8.5 | 16.1 | 0.93 | 0.528 | 6680 | 46.5 | 46.5 | 699 | 763 | 754 | 817 | 0.91 | 0.92 | DS |
| 1031 | 18 | 18 | 8.5 | 16.0 | 1.23 | 0.531 | 6590 | 47.5 | 47.5 | 922 | $\underline{4}$ | 922 | 889 | - | 1.04 | DS |

NOTES:
pes of Failure (see Figs. 13 to 16 )

$$
\begin{aligned}
& \mathrm{B} \text { - Bearine } \\
& \mathrm{CE} \text { - Corbel End, orack intersecting inclined face } \\
& \mathrm{DS} \text { - Dlagonal Splitting }
\end{aligned}
$$

FC - Flexural Compression
FT - Plexural Tension
$\ell_{*}=12 \mathrm{in}$. and $\mathrm{b}=8 \mathrm{in}$. for all specimens unless otherwise noted
not measured or inconclusive test data
For 39 specimens Avg. $\mathrm{v}_{\mathrm{y}}$ test $/ \mathrm{v}_{\mathrm{y}} \mathrm{eale}=1.06$, Standard Devtation $=0.135$,
For 78 specimens Avg. $v_{u}$ test $/ v_{u}$ calc $=1.02$, Standard Deviation $=0.119$. (Failure types $B$ and ce excluded.)
$3 \mathrm{w}=5 \mathrm{in}$. (in all other cases $w=3 \mathrm{in}$.)
$\underline{\underline{2}} \boldsymbol{f}=6 \mathrm{in}$. and $\mathrm{b}=16 \mathrm{in}$,
$3 \ell=24 \mathrm{in}$.
${ }^{4} f_{\text {su }}$ calculated swaller than $f_{y}^{\prime}$

Table A3-Test Results for Combined Load Series
( $\mathrm{H} / \mathrm{V}=\mathbf{1 / 2 )}$

| No. | $\begin{aligned} & \mathrm{h}, \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & h^{\prime}, \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \text { a, } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \mathrm{d}, \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \mathrm{p}, \\ & \% \end{aligned}$ | $\frac{a}{\bar{d}}$ | $\begin{array}{r} f_{c}^{\prime}, \\ \mathrm{psi} \\ \hline \end{array}$ |  | $\begin{aligned} & \mathrm{f}_{\mathrm{su}} \text { test, } \\ & \mathrm{ksi} \\ & \hline \end{aligned}$ | $\begin{aligned} & \mathrm{v}_{\mathrm{y}} \text { test, } \\ & \mathrm{psi} \\ & \hline \end{aligned}$ | $\begin{gathered} \mathrm{v}_{\mathrm{y}} \mathrm{ca} \\ \mathrm{psi} \end{gathered}$ | $\begin{gathered} v_{u} \text { tes } \\ \text { psi } \end{gathered}$ | $\begin{gathered} \mathrm{v}_{\mathrm{u}} \mathrm{calc}, \\ \mathrm{psi} \\ \hline \end{gathered}$ | $\frac{v_{y} \text { test }}{v_{y} \text { calc }}$ | $\frac{\mathrm{v}_{\mathrm{u}} \text { test }}{\mathrm{v}_{\mathrm{u}} \mathrm{calc}}$ | $\begin{gathered} \text { Type } \\ \text { Failures } \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 104 | 18 | 6 | 2.75 | 16.2 | 0.48 | 0.170 | 4210 | 45.7 | 45.7 | 309 | 294 | 434 | 380 | 1.05 | 1.14 | S |
| 105 | 22 | 6 | 2.75 | 20.2 | 0.38 | 0.136 | 3860 | 45.7 | 45.7 | 278 | 257 | 384 | 427 | 1.08 | 1.17 | 5 |
| 106 | 26 | 6 | 2.75 | 24.2 | 0.32 | 0.114 | 4040 | 47.3 | 47.3 | 258 | 243 | 358 | 305 | 1.06 | 1.17 | S |
| 107 | 18 | 6 | 2.75 | 16.1 | 0.93 | 0.171 | 4080 | 48.5 | 48.5 | 543 | 511 | 621 | 534 | 1.06 | 1.16 | 5 |
| 108 | 22 | 6 | 2.75 | 20.1 | 0.75 | 0.137 | 3860 | 47.7 | 47.7 | 463 | 431 | 515 | 466 | 1.07 | 1.10 | S |
| 109 | 26 | 6 | 2.75 | 24.1 | 0.62 | 0.114 | 4240 | 48.2 | 48.2 | 441 | 397 | 519 | 445 | 1.11 | 1.16 | S |
| 110 | 18 | 6 | 2.75 | 16.1 | 1.86 | 0.171 | 4250 | 47.5 | 47.5 | 932 | $\stackrel{2}{2}$ | 932 | 788 | 1.1 .2 | 1.18 | S |
| 111 | 22 | 6 | 2.75 | 20.1 | 1.49 | 0.137 | 3900 | 48.8 | 48.8 | 793 | $\underline{2}$ | 793 | 678 | - | 1.17 | S |
| 112 | 26 | 6 | 2.75 | 24.1 | 1.24 | 0.114 | 4310 | 48.7 | 48.7 | 726 | 2 | 726 | 650 | - | 1.12 | S |
| 113 | 18 | 6 | 6.0 | 16.2 | 0.47 | 0.370 | 4400 | 46.5 | 46.5 | 270 | 246 | 376 | 334 | 1.10 | 1.12 | F'r |
| 114 | 22 | 6 | 6.0 | 20.2 | 0.38 | 0.297 | 4320 | 45.7 | 45.7 | 248 | 223 | 334 | 314 | 1.11 | 1.06 | FT |
| 115 | 26 | 6 | 6.0 | 24.2 | 0.32 | 0.248 | 4950 | 45.7 | 45.7 | 207 | 215 | 339 | 318 | 0.96 | 1.06 | FT |
| 116 | 18 | 6 | 6.0 | 16.1 | 0.93 | 0.372 | 3870 | 48.3 | 48.3 | 483 | 405 | 483 | 446 | 1.19 | 1.09 | S |
| 117 | 22 | 6 | 6.0 | 20.1 | 0.75 | 0.298 | 3880 | 44.7 | 44.7 | 404 | 344 | 451 | 424 | 1.17 | 1.06 | S |
| 118 | 26 | 6 | 6.0 | 24.1 | 0.62 | 0.249 | 4240 | 48.4 | 48.4 | 363 | 343 | 454 | 419 | 1.06 | 1.08 | S |
| 119 | 18 | 6 | 6.0 | 16.1 | 1.86 | 0.372 | 4210 | 48.5 | 48.5 | 776 | 3 | 815 | 674 |  | 1.21 | S |
| 120 | 22 | 6 | 6.0 | 20.1 | 1.49 | 0.298 | 4130 | 47.7 | 47.7 | 700 | $\underline{2}$ | 715 | 634 | - | 1.13 | S |
| 121 | 26 | 6 | 6.0 | 24.1 | 1.24 | 0.249 | 3970 | 48.2 | 48.2 | 596 | $\underline{2}$ | 596 | 586 | - | 1.02 | S |
| 122 | 18 | 6 | 10.0 | 16.2 | 0.48 | 0.617 | 3380 | 46.5 | 46.5 | 174 | 185 | 211 | 234 | 0.94 | 0.90 | DS |
| 123 | 22 | 6 | 10.0 | 20.2 | 0.38 | 0.495 | 4240 | 46.5 | 46.5 | 178 | 188 | 209 | 260 | 0.94 | 0.80 | DS |
| 124 | 26 | 6 | 10.0 | 24.2 | 0.32 | 0.413 | 4240 | 46.5 | 46.5 | 155 | 181 | 207 | 255 | 0.86 | 0.81 | FT |
| 125 | 18 | 6 | 10.0 | 16.1 | 0.93 | 0.621 | 3250 | 47.9 | 31.1 | - | 301 | 236 | 326 |  | (0.72) | CE |
| 126 | 18 | 18 | 10.0 | 16.1 | 0.93 | 0.621 | 4480 | 53.4 | 53.4 | 344 | 357 | 344 | 383 | 0.96 | 0.90 | DS |
| 127 | 22 | 6 | 10.0 | 20.1 | 0.75 | 0.497 | 3300 | 47.9 | 39.6 | - | 286 | 270 | 326 | 0. | (0.83) | CE |
| 128 | 26 | 6 | 10.0 | 24.1 | 0.62 | 0.415 | 3610 | 48.0 | 39.6 | - | 277 | 244 | 334 | - | (0.73) | CE |
| 129 | 26 | 26 | 10.0. | 24.1 | 0.62 | 0.415 | 4120 | 47.0 | 47.0 | 285 | 282 | 337 | 357 | 1.01 | 0.94 | S |
| 130 | 18 | 6 | 10.0 | 16.1 | 1.86 | 0.621 | 3930 | 47.7 | 22.7 | - | $\underline{2}$ | 324 | 518 | - | (0.62) | CE |
| 131 | 18 | 18 | 10.0 | 16.1 | 1.86 | 0.621 | 4220 | 45.0 | 31.0 | - | $\underline{ }$ | 426 | 537 | - | 0.79 | DS |
| 132 | 22 | 6 | 10.0 | 20.1 | 1.49 | 0.497 | 4120 | 44.7 | 26.2 | - | 505 | 280 | 527 | - | $(0.53)$ | CE |
| 133 | 26 | 6 | 10.0 | 24.1 | 1.24 | 0.415 | 4180 | 48.4 | 35.4 | - | 506 | 279 | 521 | - | $(0.54)$ | CE |
| 134 | 26 | 26 | 10.0 | 24.1 | 1.24 | 0.415 | 4290 | 43.9 | 38.4 | - | 469 | 458 | 528 | - | 0.87 | DS |
|  |  |  |  |  |  |  |  | For 17 specimens Avg. $v_{y}$ test $/ v_{y} c a l c=1.04$, Standard Deviation $=0.088$; For 25 specimens Avg. $v_{u}$ test $/ v_{u} c a l c=1.05$, Standard Deviation $=0.132$. |  |  |  |  |  |  |  |  |

# Table A3-Test Results for Combined Load Series (continued) 

( $H / V=1 / 1$ )

| No. | $\begin{aligned} & \text { h, } \\ & \text { in. } \end{aligned}$ |  | $\begin{gathered} \text { a. } \\ \text { in. } \end{gathered}$ | d, in. | $\begin{aligned} & \mathrm{p}, \\ & \% \end{aligned}$ | $\frac{a}{d}$ | $f_{c}^{\prime}$, <br> psi | $\mathrm{f}_{\mathrm{y}}$ | $\begin{gathered} f_{\text {sust }} \text { test } \\ \mathrm{ksi} \\ \hline \end{gathered}$ | $\begin{aligned} & \mathrm{v}_{\mathrm{y}} \text { test, } \\ & \mathrm{psi} \end{aligned}$ | $\begin{aligned} & \mathrm{v}_{\mathrm{y}} \mathrm{ca} \\ & \mathrm{psi} \end{aligned}$ | $\mathrm{v}_{\mathrm{u}}$ te <br> psi | $\mathrm{v}_{\mathrm{u}} \mathrm{cale}$, psi | $\frac{v_{\text {y test }}}{v_{y} \text { calc }}$ | $\frac{v_{u} \text { test }}{v_{u} \operatorname{calc}}$ | Type Failures |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 136 | 18 | 6 | 2.75 | 16.2 | 0.48 | 0.170 | 3870 | 47.0 | 47.0 | 174 | $\underline{\square}$ | 251 | 198 | - | 1.26 | FT |
| 137 | 22 | 6 | 2.75 | 20.2 | 0.38 | 0.136 | 4610 | 44.3 | 44.3 | 155 | 185 | 224 | 186 | 0.84 | 1.20 | FT |
| 138 | 26 | 6 | 2.75 | 24.2 | 0.32 | 0.114 | 3870 | 46.8 | 46.8 | 134 | $\underline{2}$ | 213 | 150 |  | 1.42 | FT |
| 139 | 18 | 6 | 2.75 | 16.1 | 0.93 | 0.171 | 4420 | 44.3 | 44.3 | 373 | 343 | 503 | 346 | 1.08 | 1.45 | FT |
| 140 | 22 | 6 | 2.75 | 20.1 | 0.75 | 0.137 | 3890 | 44.3 | 44.3 | 295 | $\underline{2}$ | 373 | 278 | - | 1.34 | FT |
| 141 | 26 | 6 | 2.75 | 24.1 | 0.62 | 0.114 | 4000 | 45.3 | 45.3 | 233 | 2 | 337 | 248 | - | 1.36 | S |
| 142 | 18 | 6 | 2.75 | 16.1 | 1.86 | 0.171 | 4270 | 44.3 | 44.3 | 660 | $\underline{2}$ | 660 | 564 | - | 1.17 | S |
| 143 | 22 | 6 | 2.75 | 20.1 | 1.49 | 0.137 | 4110 | 47.2 | 47.2 | 553 | 2 | 575 | 476 | - | 1.21 | S |
| 144 | 26 | 6 | 2.75 | 24.1 | 1.24 | 0.114 | 4250 | 48.8 | 48.8 | 519 | $\underline{3}$ | 532 | 425 | - | 1.25 | 5 |
| 145 | 18 | 6 | 6.0 | 16.2 | 0.48 | 0.370 | 3720 | 45.0 | 45.0 | 154 | 166 | 212 | 167 | 0.93 | 1.27 | FT |
| 146 | 22 | 6 | 6.0 | 20.2 | 0.38 | 0.297 | 4300 | 45.0 | 45.0 | 136 | 157 | 166 | 163 | 0.87 | 1.02 | FT |
| 147 | 26 | 6 | 6.0 | 24.2 | 0.32 | 0.248 | 4040 | 45.0 | 45.0 | 110 | $\underline{\underline{2}}$ | 245 | 144 | - | 1.70 | FT |
| 148 | 18 | 6 | 6.0 | 16.1 | 0.93 | 0.372 | 4250 | 43.6 | 43.6 | 272 | 272 | 388 | 291 | 1.00 | 1.33 | FT |
| 149 | 22 | 6 | 6.0 | 20.1 | 0.75 | 0.298 | 4320 | 43.6 | 43.6 | 230 | 247 | 349 | 266 | 0.93 | 1.31 | FT |
| 150 | 26 | 6 | 6.0 | 24.1 | 0.62 | 0.249 | 4050 | 43.6 | 43.6 | 207 | 223 | 311 | 235 | 0.93 | 1.32 | FT |
| 151 | 18 | 6 | 6.0 | 16.1 | 1.86 | 0.372 | 4230 | 45.3 | 45.3 | 543 | $\underline{2}$ | 543 | 492 | - | 1.10 | S |
| 152 | $22^{\circ}$ | 6 | 6.0 | 20.1 | 1.49 | 0.298 | 4130 | 48.5 | 48.5 | 482 | $\underline{2}$ | 513 | 433 | - | 1.18 | S |
| 153 | 26 | 6 | 6.0 | 24.1 | 1.24 | 0.249 | 3960 | 45.3 | 45.3 | 404 | $\underline{2}$ | 404 | 386 | 8. ${ }^{*}$ | 1.05 | DS |
| 154 | 18 | 6 | 10.0 | 16.2 | 0.48 | 0.617 | 4750 | 48.5 | 48.5 | 127 | 146 | 127 | 151 | $0.87 *$ | (0.84) | CE |
| 155 | 22 | 6 | 10.0 | 20.2 | 0.38 | 0.495 | 4120 | 48.5 | 48.5 | 121 | $\underline{2}$ | 121 | 133 | - | (0.90) | CE |
| 156 | 26 | 6 | 10.0 | 24.2 | 0.32 | 0.413 | 3670 | 48.5 | 48.5 | 110 | $\underline{2}$ | 110 | 119 |  | (0.92) | CE |
| 157 | 18 | 6 | 10.0 | 16.1 | 0.93 | 0.621 | 4150 | 45.3 | 45.3 | 194 | 218 | 194 | 229 | $0.89 *$ | (0.85) | CE |
| 158 | 18 | 9 | 10.0 | 16.1 | 0.93 | 0.621 | 4300 | 45.6 | 41.4 | - | 221 | 203 | 233 | - | (0.87) | CE |
| 159 | 18 | 18 | 10.0 | 16.1 | 0.93 | 0.621 | 4540 | 45.5 | 45.5 | 202 | 223 | 202 | 24.0 | 0.91 * | 0.84 | DS |
| 160 | 22 | 6 | 10.0 | 20.1 | 0.75 | 0.497 | 4200 | 45.3 | 45.3 | 155 | 208 | 155 | 219 | 0.75 * | (0.71) | CE |
| 161 | 26 | 6 | 10.0 | 24.1 | 0.62 | 0.415 | 4090 | 45.3 | 45.3 | 130 | 195 | 142 | 204 | 0.66 * | (0.70) | CE |
| 162 | 26 | 11 | 10.0 | 24.1 | 0.62 | 0.415 | 4470 | 43.2 | 36.9 | - | 192 | 156 | 214 | - | (0.73) | CE |
| 163 | 26 | 26 | 10.0 | 24.1 | 0.62 | 0.415 | 4350 | 46.7 | 46.7 | 182 | 202 | 182 | 211 | 0.90 | (0.86) | B |
| 164 | 18 | 6 | 10.0 | 16.1 | 1.86 | 0.621 | 4080 | 48.3 | 38.2 | - | $\underline{2}$ | 272 | 378 | . | (0.72) | CE |
| 165 | 18 | 18 | 10.0 | 16.1 | 1.86 | 0.621 | 4520 | 45.4 | 38.4 | - | 396 | 337 | 397 |  | 0.85 | DS |
| 166 | 22 | 6 | 10.0 | 20.1 | 1.49 | 0.497 | 4210 | 42.5 | 42.5 | 212 | 344 | 212 | 360 | $0.62^{*}$ | (0.59) | CE |
| 167 | 26 | 6 | 10.0 | 24.1 | 1.24 | 0.415 | 4440 | 42.5 | 39.6 |  | 326 | 192 | 554 | - | (0.54) | CE |
| 168 | 26 | 26 | 10:0 | 24.1 | 1.24 | 0.415 | 4550 | 46.7 | 46.7 | 272 | 355 | 295 | 358 | 0.77 | 0.82 | DS |
| For 10 specimens Avg. $\mathrm{v}_{\mathrm{y}}$ test/ $\mathrm{v}_{\mathrm{y}}$ calc $=0.92$, Standard Deviation $=0.084$; <br> For 21 specimens Avg. $v_{u}$ test $/ v_{u}$ calc $=1.21$, Standard Deviation $=0.216$. |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |

Table A3-Test Results for Combined Load Series (concluded)

| No. | $\begin{aligned} & \text { h, } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \text { h', } \\ & \text { in. } \end{aligned}$ | $\begin{array}{r} \text { a, } \\ \text { in. } \end{array}$ | $\begin{array}{r} \text { d, } \\ \text { in. } \end{array}$ | $\begin{aligned} & \mathrm{p}, \\ & \% \end{aligned}$ | $\frac{a}{d}$ | $\begin{aligned} & f_{c}^{\prime} \\ & \text { psi } \end{aligned}$ | $\begin{aligned} & f_{y}^{\prime} \\ & k s i \end{aligned}$ | $\begin{gathered} \mathrm{f}_{\mathrm{su}} \mathrm{tes}^{-} \\ \mathrm{ksi} \end{gathered}$ |  |  | $\mathrm{v}_{\mathrm{u}} \mathrm{te}^{\text {a }}$ psi |  | $\frac{\mathrm{v}_{\mathrm{y}} \mathrm{y}^{\mathrm{te}}{ }^{\text {cea }}}{}$ | $\frac{v_{u} \text { test }}{v_{u} \text { calc }}$ | $\begin{gathered} \text { Type } \\ \text { Failures } \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $(\mathrm{H} / \mathrm{V}=3 / 4)$ |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 1351 | 24 | 8 | 3.0 | 12.1 | 1.24 | 0.248 | 6430 | 46.8 | 46.8 | 542 | 533 | 775 | 605 | 1.02 | 1.28 | S |
| ( $\mathrm{H} / \mathrm{V}=5 / 4$ ) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 1691 | 14 | 8 | 3.0 | 12.1 | 2.48 | 0.248 | 6650 | 46.8 | 46.8 | 692 | $\underline{2}$ | 983 | 722 | - | 1.36 | S |

NOTES:
Types of Failure (see Figs. 13 to 16)
B - Bearing
CE - Corbel End, crack intersecting inclined face
DS - Diagonal Splitting
FC - Flexural Compression
FT - Flexural Tension
S - Shear
$\ell=12$ in. and $b=8 \mathrm{in}$. for all specimens unless otherwise noted

* not measured or inconclusive test data
$\underline{x} h=6$ in.
$\underline{Z}_{f_{\text {su }}}$ calculated smaller than $f_{y}$


## Table A4-Test Results for Corbels with Stirrups


$l=12 \mathrm{in}$. and $\mathrm{b}=8 \mathrm{in}$. for all specimens
$1 \mathrm{w}=15 \mathrm{in}$. ( $\mathrm{w}=3 \mathrm{in}$. for all others)
$\underset{ }{2}$ Test stopped at $v=1190 \mathrm{psi}$
3 Stirrups not included in $p$
$4 f_{\text {su }}$ calculated smaller than $f_{y}$

Table A5-Comparison with Test Results of Other Investigators

|  | Source | No. | $\begin{aligned} & \text { a, } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \text { a, } \\ & \text { in. } \end{aligned}$ | $\begin{aligned} & \mathrm{p}, \\ & \% \end{aligned}$ | $\frac{\mathrm{a}}{\mathrm{d}}$ | $\begin{aligned} & f_{c}^{\prime} \\ & p s i \\ & \hline \end{aligned}$ | $\underset{\mathrm{f}_{\mathrm{si}}}{ }$ | $\begin{gathered} \mathrm{f}_{\text {su }} \text { test, } \\ \mathrm{ksi} \end{gathered}$ | $\begin{aligned} & \mathrm{v}_{\mathrm{y}} \mathrm{test}, \\ & \mathrm{psi} \\ & \hline \end{aligned}$ | $\begin{gathered} v_{y} \mathrm{ca} \\ \mathrm{psi} \end{gathered}$ | $\mathrm{v}_{\mathrm{u}}$ tes <br> psi | $\mathrm{v}_{\mathrm{u}} \mathrm{calc}$ <br> psi | $\frac{v_{y} \text { test }}{v_{y} \text { calc }}$ | $\frac{v_{u} \text { test }}{v_{u} \text { calc }}$ | b |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | U. of I. ${ }^{(4)}$ | B-8 | 14.0 | 29.0 | 1.03 | 0.483 | 3390 | 46.0 | * | * | 626 | 621 | 628 | - | 0.99 | 4 |
|  |  | B-2-1 | 14.0 | 22.0 | 1.00 | 0.636 | 2910 | 39.9 | * | * | 440 | 511 | 501 | - | 1.02 | 4 |
|  |  | B-2-2 | 14.0 | 22.0 | 2.00 | 0.636 | 2290 | 45.9 | * | * | $\underline{1}$ | 502 | 560 | - | 0.90 | 4 |
|  |  | B-3-1 | 14.0 | 15.5 | 1.00 | 0.903 | 3740 | 50.9 | 50.9 | 403 | $\underline{1}$ | 427 | 459 | - | 0.93 | 4 |
|  |  | B-3-2 | 14.0 | 15.5 | 1.00 | 0.903 | 4940 | 51.4 | 51.4 | 516 | 1 | 532 | 527 | - | 1.01 | 4 |
|  |  | B-3-3 | 14.0 | 15.5 | 2.00 | 0.903 | 2960 | 42.4 | * | * | 1 | 645 | 514 |  | 1.25 | 4 |
|  |  | B-4-1 | 14.0 | 10.0 | 1.00 | 1.400 | 2800 | 45.0 | 45.0 | 237 | 261 | 237 | 289 | 0.91 | 0.82 | 4 |
|  |  | B-4-2 | 14.0 | 10.0 | 2.20 | 1.400 | 2520 | 54.5 | * | * | 1 | 317 | 357 | - | 0.89 | 4 |
|  |  | B-4-3 | 14.0 | 10.0 | 1.00 | 1.400 | 6460 | 47.3 | 47.3 | 285 | 404 | 372 | 439 | 0.70 | 0.85 | 4 |
|  |  | Avg. $\mathrm{v}_{\mathrm{u}}$ test $/ \mathrm{v}_{\mathrm{u}} \mathrm{calc}=0.96 ;$ Standard Deviation $=0.128$ |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  | U. of I. ${ }^{(5)}$ | F4S1 | 6.0 | 6.0 | 0.83 | 1.000 | 4970 | 46.7 | 46.7 | 306 | 411 | 442 | 464 | 0.74 | 0.95 | 4 |
|  |  | F4S2 | 6.0 | 6.0 | 1.67 | 1.000 | 5030 | 48.6 | 48.6 | 614 | 1 | 854 | 589 | $\cdots$ | 1.45 | 4 |
|  |  | F352 | 6.0 | 8.0 | 0.83 | 0.750 | 3530 | 47.4 | 47.4 | 427 | 433 | 575 | 472 | 0.98 | 1.22 | 3 |
|  |  | F3S3 | 6.0 | 8.0 | 1.67 | 0.750 | 4980 | 47.4 | 47.4 | 700 | 2 | 1140 | 707 | - | 1.61 | 3 |
|  |  | F2Sl | 6.0 | 12.0 | 0.83 | 0.500 | 4920 | 46.0 | 46.0 | 610 | 616 | 902 | 693 | 0.99 | 1.30 | 2 |
|  |  | F2S2 | 6.0 | 12.0 | 1.29 | 0.500 | 4600 | 44.8 | 44.8 | 885 | 1 | 1150 | 844 | O. | 1.36 | 2 |
|  |  | Specimens have compression reinforcement |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  | U. of I. ${ }^{(6)}$ | G23S-11 | 6.0 | 13.0 | 0.83 | 0.462 | 3560 | 45.7 | 45.7 | 533 | 516 | 776 | 595 | 1.03 | 1.30 | 2 |
|  |  | G23S-21 | 6.0 | 13.0 | 0.46 | 0.462 | 3420 | 51.4 | 51.4 | 325 | 362 | 462 | 478 | 0.90 | 0.97 | 2 |
|  |  | G24S-11 | 6.0 | 13.0 | 0.83 | 0.462 | 5600 | 45.7 | 45.7 | 535 | 636 | 785 | 746 | 0.84 | 1.05 | 2 |
|  |  | G24S-21 | 6.0 | 13.0 | 0.46 | 0.462 | 5240 | 51.4 | 51.4 | 323 | 437 | 435 | 591 | 0.74 | 0.74 | 2 |
|  |  | G335-11 | 6.0 | 9.0 | 1.67 | 0.667 | 3380 | 47.3 | 47.3 | 667 | 1 | 711 | 600 | 0.7 | 1.18 | 3 |
|  |  | G335-21 | 6.0 | 9.0 | 0.83 | 0.667 | 3050 | 45.2 | 45.2 | 324 | 382 | 454 | 452 | 0.85 | 1.00 | 3 |
|  |  | G335-31 | 6.0 | 9.0 | 2.58 | 0.667 | 2890 | 45.2 | 45.2 | 861 | $\underline{1}$ | 891 | 642 | . | 1.39 | 3 |
|  |  | G345-11 | 6.0 | 9.0 | 1.67 | 0.667 | 5100 | 47.2 | 47.2 | 694 | 1 | 915 | 737 | - | 1.24 | 3 |
|  |  | G34S-21 | 6.0 | 9.0 | 0.83 | 0.667 | 4960 | 47.0 | 47.0 | 359 | 493 | 467 | 577 | 0.73 | 0.81 | 3 |
|  |  | G43S-11 | 6.0 | 7.0 | 1.67 | 0.857 | 3510 | 44.1 | 44.1 | 475 | $\underline{1}$ | 618 | 518 | - | 1.19 | 4 |
|  |  | G44S-11 | 6.0 | 7.0 | 1.67 | 0.857 | 3560 | 47.9 | 47.9 | 530 | 1 | 671 | 522 | - | 1.29 | 4 |
|  |  | Avg. $\mathrm{v}_{\mathrm{u}}$ test $/ \mathrm{v}_{\mathrm{u}}$ calc $=1.10 ;$ Standard Deviation $=0.209$ |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  | U. of I. ${ }^{(7)}$ | HOa | 6.0 | 8.0 | 0.83 | 0.750 | 2930 | 45.0 | 45.0 | 338 | 381 | 367 | 430 | 0.89 | 0.85 | 3 |
|  |  | HOb | 6.0 | 8.0 | 0.83 | 0.750 | 5800 | 51.0 | 51.0 | 421 | 581 | 448 | 605 | 0.72 | 0.74 | 3 |
|  |  | HOn | 6.0 | 8.0 | 0.83 | 0.750 | 3580 | 51.0 | 51.0 | 351 | 463 | 382 | 476 | 0.76 | 0.80 | 3 |
|  |  | $\text { Avg. } v_{u} \text { test } / v_{u} \text { calc }=0.80$ |  |  |  |  |  |  |  |  |  |  |  |  |  |  |

Table A5-Comparison with Test Results of Other Investigators (concluded)


* Not measured or inconclusive test data
$\underline{1} f_{\text {su }}$ calculated smaller than $f_{y}$

Table A6-Corbel Bearing Stresses at Ultimate Strength

| No. | $\begin{aligned} & f_{c}^{\prime} \\ & p s i \\ & \hline \end{aligned}$ | $\begin{aligned} & f_{b u}, \\ & \text { psi } \end{aligned}$ | $\frac{\mathrm{f}_{\mathrm{bu}}}{\mathrm{f}_{\mathrm{c}}^{\mathrm{f}}}$ | Type Failures | No. | $\begin{aligned} & \mathrm{f}_{\mathrm{c}}^{\prime} \\ & \mathrm{psi} \\ & \hline \end{aligned}$ | $\begin{aligned} & \mathrm{f}_{\text {bu }}, \\ & \text { psi. } \end{aligned}$ | $\begin{aligned} & \frac{f_{b u}}{f_{c}^{\prime}} \end{aligned}$ | Type <br> Failures | No. | $\begin{aligned} & \mathrm{f}_{\mathrm{c}}, \\ & \mathrm{psi} \end{aligned}$ | $\begin{aligned} & \mathrm{f}_{\mathrm{bu}}, \\ & \mathrm{psi} \\ & \hline \end{aligned}$ | $\frac{f_{b u}}{f_{c}^{\prime}}$ | Type <br> Failures |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 3790 | 4170 | 1.10 | S | 41 | 4200 | 1950 | 0.46 | DS | 81 | 2570 | 2700 | 1.05 | FC |
| 2 | 6170 | 4410 | 0.71 | S | 42 | 4850 | 3750 | 0.77 | S | 82 | 2110 | 2110 | 1.00 | FC |
| 3 | 3820 | 4540 | 1.19 | S | 43 | 4140 | 3000 | 0.72 | S | 83 | 2310 | 1290 | 0.56 | FC |
| 4 | 3520 | 4170 | 1.18 | S | 44 | 3840 | 2800 | 0.73 | DS | 84 | 2290 | 1750 | 0.76 | FC |
| 5 | 3840 | 5690 | 1.48 | S | 45 | 4280 | 3000 | 0.70 | DS | 85 | 2170 | 1580 | 0.73 | FC |
| 6 | 3970 | 5730 | 1.44 | S | 46 | 3840 | 2620 | 0.68 | DS | 86 | 4180 | 4250 | 1.02 | S |
| 7 | 3260 | 5860 | 1.80 | 5 | 47 | 4060 | 2610 | 0.64 | DS | 87 | 3880 | 3880 | 1.00 | S |
| 8 | 4170 | 4370 | 1.05 | S | 48 | 4920 | 3460 | 0.70 | CE | 88 | 3820 | 3730 | 0.98 | S |
| 9 | 6500 | 6640 | 1.02 | S | 49 | 4180 | 3450 | 0.82 | B | 89 | 4010 | 2750 | 0.68 | FC |
| 10 | 4790 | 7220 | 1.51 | 5 | 50 | 4390 | 2300 | 0.52 | B | 90 | 4240 | 3890 | 0.92 | 5 |
| 11 | 3900 | 3850 | 0.99 | DS | 51 | 4490 | 2710 | 0.60 | B | 91 | 4060 | 3080 | 0.76 | S |
| 12 | 4240 | 2940 | 0.69 | FT | 52 | 3960 | 2000 | 0.50 | B | 92 | 4160 | 1610 | 0.39 | DS |
| 13 | 4580 | 4000 | 0.87 | FT | 53 | 6360 | 2270 | 0.36 | B | 93 | 3980 | 2250 | 0.56 | DS |
| 14 | 4540 | 3500 | 0.77 | DS | 54 | 3950 | 1880 | 0.47 | B | 94 | 3940 | 2840 | 0.72 | DS |
| 15 | 4500 | 3000 | 0.67 | DS | 55 | 4010 | 2520 | 0.63 | CE | 95 | 6310 | 4750 | 0.75 | S |
| 16 | 3430 | 1960 | 0.57 | DS | 56 | 3770 | 2420 | 0.64 | CE | 96 | 6430 | 6260 | 0.97 | S |
| 17 | 3990 | 3560 | 0.89 | DS | 57 | 4130 | 3500 | 0.85 | B | 97 | 6420 | 5350 | 0.83 | S |
| 18 | 4210 | 3380 | 0.80 | S | 58 | 3720 | 2330 | 0.63 | CE | 98 | 6610 | 3180 | 0.48 | S |
| 19 | 3790 | 3850 | 1.02 | DS | 59 | 3510 | 2270 | 0.65 | CE | 99 | 6570 | 4610 | 0.70 | DS |
| 20 | 3550 | 3590 | 1.01 | DS | 60 | 3820 | 3330 | 0.87 | FC | 100 | 6430 | 4280 | 0.66 | S |
| 21 | 3920 | 3960 | 1.01 | DS | 61. | 4110 | 3130 | 0.76 | DS | 102 | 6370 | 1950 | 0.31 | FC |
| 22 | 3740 | 4370 | 1.17 | DS | 62 | 3260 | 2920 | 0.90 | CE | 102 | 6680 | 2430 | 0.36 | DS |
| 23 | 3950 | 3680 | 0.93 | DS | 63 | 3420 | 2720 | 0.80 | CE | 103 | 6590 | 2950 | 0.45 | DS |
| 24 | 4250 | 3710 | 0.87 | DS | 64 | 6540 | 3960 | 0.60 | CE | 104 | 4210 | 2340 | 0.56 | S |
| 25 | 6410 | 5420 | 0.84 | DS | 65 | 3660 | 4170 | 1.14 | B | 105 | 3860 | 2590 | 0.67 | S |
| 26 | 4280 | 4610 | 1.08 | S | 66 | 4040 | 3860 | 0.96 | B | 106 | 4040 | 2890 | 0.72 | S |
| 27 | 4320 | 4790 | 1.11 | DS | 67 | 4060 | 3680 | 0.91 | B | 107 | 4080 | 3330 | 0.82 | S |
| 28 | 4630 | 5210 | 1.12 | DS | 68 | 3380 | 2240 | 0.66 | CE | 108 | 3860 | 3450 | 0.89 | S |
| 29 | 3730 | 5140 | 1.38 | S | 69 | 3680 | 3590 | 0.98 | DS | 109 | 4240 | 4170 | 0.98 | S |
| 30 | 4260 | 5620 | 1.32 | DS | 70 | 4010 | 3330 | 0.83 | S | 110 | 4250 | 5000 | 1.18 | S |
| 31 | 4040 | 3750 | 0.93 | DS | 71 | 4410 | 3290 | 0.75 | CE | 111 | 3900 | 5310 | 1.36 | 5 |
| 32 | 4390 | 5830 | 1.33 | DS | 72 | 4110 | 3150 | 0.77 | CE | 112 | 4310 | 5830 | 1.35 | S |
| 33 34 | 3830 | 4750 | 1.24 | DS | 73 | 4050 | 4580 | 1.13 | B | 113 | 4400 | 2030 | 0.46 | FT |
| 34 35 | 4070 3820 | 5140 | 1.26 | DS | 74 | 4360 | 4810 | 1.10 | B | 214 | 4320 | 2250 | 0.52 | FT |
| 35 | 3820 | 5510 | 1.44 | B | 75 | 4110 | 2680 | 0.65 | DS | 115 | 4950 | 2730 | 0.55 | FT |
| 36 | 3960 | 6460 | 1.63 | DS | 76 | 4090 | 3120 | 0.76 | DS | 116 | 3870 | 2590 | 0.67 | S |
| 37 38 | 3770 | 6500 | 1.72 | S | 77 | 2210 | 2300 | 1.04 | S | 117 | 3880 | 3020 | 0.78 | 5 |
| 38 | 4700 | 2140 | 0.46 | DS | 78 | 2200 | 2630 | 1.20 | S | 118 | 4240 | 3640 | 0.86 | S |
| 39 | 4490 | 2170 | 0.48 | DS | 79 | 2400 | 2490 | 1.04 | B | 119 | 4210 | 4380 | 1.04 | S |
| 40 | 4340 | 2180 | 0.50 | FC | 80 | 2430 | 2080 | 0.86 | FC | 120 | 4130 | 4790 | 1.16 | S |

Table A6-Corbel Bearing Stresses at Ultimate Strength (concluded)

| No. | $f_{c}^{\prime}$, <br> psi | $\begin{aligned} & f_{b u} \\ & p s i \end{aligned}$ | $\begin{aligned} & \mathrm{f}_{\mathrm{bu}} \\ & \mathrm{f}_{\mathrm{c}}^{1} \end{aligned}$ | Type <br> Failure | No. | $\begin{aligned} & f_{c}^{\prime} \\ & p s i \\ & \hline \end{aligned}$ | $\begin{aligned} & f_{\text {bu }}, \\ & \text { psi } \end{aligned}$ | $\frac{f_{b_{u}}}{f_{c}^{\prime}}$ | Type <br> Failure | No. | $\begin{aligned} & f_{c}^{\prime} \\ & p: 1 \end{aligned}$ | $\begin{aligned} & \mathbf{f}_{\mathrm{bu}}, \\ & \mathrm{psi} \\ & \hline \end{aligned}$ | $\frac{\mathrm{f}_{\mathrm{bu}}}{\mathrm{f}_{c}^{1}}$ | Type Failure |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 121 | 3970 | 4790 | 1.21 | 5 | 138 | 3870 | 1720 | 0.44 | FT | 155 | 41.20 | 810 | 0.20 | CE |
| 122 | 3380 | 1140 | 0.34 | DS | 139 | 4420 | 2700 | 0.62 | FT | 156 | 3670 | 880 | 0.24 | CE |
| 123 | 4240 | 1410 | 0.33 | DS | 140 | 3890 | 2500 | 0.64 | FT | 157 | 4150 | 1040 | 0.25 | CE |
| 124 | 4240 | 1670 | 0.39 | FT | 141 | 4000 | 2710 | 0.68 | 5 | 158 | 4300 | 1090 | 0.25 | CE |
| 125 | 3250 | 1260 | 0.39 | CE | 142 | 4270 | 3540 | 0.83 | 5 | 159 | 4540 | 1080 | 0.24 | DS |
| 126 | 4480 | 1840 | 0.41 | DS | 143 | 4110 | 3850 | 0.94 | S | 160 | 4200 | 1040 | 0.25 | CE |
| 127 | 3300 | 1810 | 0.55 | CE | 144 | 4250 | 4260 | 1.00 | 5 | 161 | 4090 | 1140 | 0.28 | CE |
| 128 | 3610 | 1960 | 0.54 | CE | 145 | 3720 | 1140 | 0.31 | FT | 162 | 4470 | 1250 | 0.28 | CE |
| 129 | 4120 | 2710 | 0.66 | S | 146 | 4300 | 1120 | 0.26 | FT | 163 | 4350 | 1460 | 0.34 | B |
| 130 | 3930 | 1740 | 0.44 | CE | 147 | 4040 | 1980 | 0.49 | FT | 164 | 4080 | 1460 | 0.36 | CE |
| 131 | 4220 | 2290 | 0.54 | DS | 148 | 4250 | 2080 | 0.49 | FT | 165 | 4520 | 1810 | 0.40 | DS |
| 132 | 4120 | 1880 | 0.46 | CE | 149 | 4320 | 2340 | 0.54 | FT | 166 | 4110 | 1420 | 0.34 | CE |
| 133 | 4180 | 2240 | 0.54 | CE | 150 | 4050 | 2500 | 0.62 | FT | 167 | 4440 | 1540 | 0.35 | CE |
| 134 | 4290 | 3680 | 0.86 | DS | 151 | 4230 | 2920 | 0.69 | 5 | 168 | 4550 | 2370 | 0.52 | DS |
| 135. | 6430 | 1560 | 0.24 | S | 152 | 4130 | 3440 | 0.83 | S | 169 | 6650 | 1980 | 0.30 | S |
| 136 | 3870 | 1350 | 0.35 | FT | 153 | 3960 | 3250 | 0.82 | DS |  |  |  |  |  |
| 137 | 4610 | 1510 | 0.33 | FT | 154 | 4750 | 690 | 0.14 | CE |  |  |  |  |  |

$$
\begin{aligned}
& \mathrm{H} / \mathrm{V}=0 \text { for } 1 \text { to } 103 \\
& \mathrm{H} / \mathrm{V}=1 / 2 \text { for } 104 \text { to } 134 \\
& \mathrm{H} / \mathrm{V}=3 / 4 \text { for } 135 \\
& \mathrm{H} / \mathrm{V}=1 / 1 \text { for } 136 \text { to } 168 \\
& H / V=5 / 4 \text { for } 169
\end{aligned}
$$

Table A7-Values of $\mathrm{C}_{2}=0.8(10)^{\mathrm{a} / 3 \mathrm{~d}}$ ( $\mathrm{H} / \mathrm{V}=0$ and no stirrups)

| $a / d$ | 0.00 | 0.01 | 0.02 | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 | 0.08 | 0.09 |
| :---: | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 0.0 | 0.80 | 0.81 | 0.81 | 0.82 | 0.82 | 0.83 | 0.84 | 0.84 | 0.85 | 0.86 |
| 0.1 | 0.86 | 0.87 | 0.88 | 0.88 | 0.89 | 0.90 | 0.90 | 0.91 | 0.92 | 0.92 |
| 0.2 | 0.93 | 0.94 | 0.95 | 0.95 | 0.96 | 0.97 | 0.98 | 0.98 | 0.99 | 1.00 |
| 0.3 | 1.01 | 1.01 | 1.02 | 1.03 | 1.04 | 1.05 | 1.05 | 1.06 | 1.07 | 1.08 |
| 0.4 | 1.09 | 1.10 | 1.10 | 1.11 | 1.12 | 1.13 | 1.14 | 1.15 | 1.16 | 1.16 |
| 0.5 | 1.17 | 1.18 | 1.19 | 1.20 | 1.21 | 1.22 | 1.23 | 1.24 | 1.25 | 1.26 |
| 0.6 | 1.27 | 1.28 | 1.29 | 1.30 | 1.31 | 1.32 | 1.33 | 1.34 | 1.35 | 1.36 |
| 0.7 | 1.37 | 1.38 | 1.39 | 1.40 | 1.41 | 1.42 | 1.43 | 1.44 | 1.45 | 1.47 |
| 0.8 | 1.48 | 1.49 | 1.50 | 1.51 | 1.52 | 1.54 | 1.55 | 1.56 | 1.57 | 1.58 |
| 0.9 | 1.60 | 1.61 | 1.62 | 1.63 | 1.64 | 1.66 | 1.67 | 1.68 | 1.70 | 1.71 |

Table A8-Values of $\mathrm{C}_{2}=0.25(10)^{\mathrm{a} / \mathrm{d}}$

| $a / d$ | 0.00 | 0.01 | 0.02 | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 | 0.08 | 0.09 |
| :---: | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 0.0 | 0.25 | 0.26 | 0.26 | 0.27 | 0.27 | 0.28 | 0.29 | 0.29 | 0.30 | 0.31 |
| 0.1 | 0.31 | 0.32 | 0.33 | 0.34 | 0.34 | 0.35 | 0.36 | 0.37 | 0.38 | 0.39 |
| 0.2 | 0.40 | 0.40 | 0.41 | 0.42 | 0.43 | 0.44 | 0.45 | 0.46 | 0.48 | 0.49 |
| 0.3 | 0.50 | 0.51 | 0.52 | 0.53 | 0.55 | 0.56 | 0.57 | 0.59 | 0.60 | 0.61 |
| 0.4 | 0.63 | 0.64 | 0.66 | 0.67 | 0.69 | 0.70 | 0.72 | 0.74 | 0.75 | 0.77 |
| 0.5 | 0.79 | 0.81 | 0.83 | 0.85 | 0.87 | 0.89 | 0.91 | 0.93 | 0.95 | 0.97 |
| 0.6 | 1.00 | 1.02 | 1.04 | 1.07 | 1.09 | 1.12 | 1.14 | 1.17 | 1.20 | 1.22 |
| 0.7 | 1.25 | 1.28 | 1.31 | 1.34 | 1.37 | 1.40 | 1.44 | 1.47 | 1.51 | 1.54 |
| 0.8 | 1.58 | 1.61 | 1.65 | 1.69 | 1.73 | 1.77 | 1.81 | 1.85 | 1.90 | 1.94 |
| 0.9 | 1.98 | 2.03 | 2.08 | 2.13 | 2.18 | 2.23 | 2.28 | 2.33 | 2.39 | 2.44 |

Table A9-Values of $C_{1}=1.5(\mathrm{a} / \mathrm{d})^{2 / 3}$
( $\mathrm{H} / \mathrm{V}$ does not equal O )

| $\mathrm{a} / \mathrm{d}$ | 0.00 | 0.01 | 0.02 | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 | 0.08 | 0.09 |
| :---: | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 0.0 | 0.00 | 0.07 | 0.11 | 0.14 | 0.18 | 0.20 | 0.23 | 0.25 | 0.28 | 0.30 |
| 0.1 | 0.34 | 0.36 | 0.38 | 0.40 | 0.42 | 0.44 | 0.46 | 0.48 | 0.50 | 0.51 |
| 0.2 | 0.53 | 0.55 | 0.56 | 0.58 | 0.60 | 0.61 | 0.63 | 0.64 | 0.66 | 0.67 |
| 0.3 | 0.69 | 0.70 | 0.72 | 0.73 | 0.74 | 0.76 | 0.77 | 0.79 | 0.80 | 0.81 |
| 0.4 | 0.83 | 0.84 | 0.85 | 0.87 | 0.88 | 0.89 | 0.91 | 0.92 | 0.93 | 0.94 |
| 0.5 | 0.96 | 0.97 | 0.98 | 0.99 | 1.01 | 1.02 | 1.03 | 1.04 | 1.06 | 1.07 |
| 0.6 | 1.08 | 1.09 | 1.10 | 1.11 | 1.12 | 1.14 | 1.15 | 1.16 | 1.17 | 1.18 |
| 0.7 | 1.19 | 1.20 | 1.22 | 1.23 | 1.24 | 1.25 | 1.26 | 1.27 | 1.28 | 1.29 |
| 0.8 | 1.30 | 1.31 | 1.32 | 1.34 | 1.35 | 1.36 | 1.37 | 1.38 | 1.39 | 1.40 |
| 0.9 | 1.41 | 1.42 | 1.43 | 1.44 | 1.45 | 1.46 | 1.47 | 1.48 | 1.49 | 1.50 |

Table A10-Values of $C_{2}=0.7(10)^{\mathrm{a} / 2 \mathrm{~d}}$
$(\mathrm{H} / \mathrm{V}$ does not equal 0$)$

| $\mathrm{a} / \mathrm{d}$ | 0.00 | 0.01 | 0.02 | 0.03 | 0.04 | 0.05 | 0.06 | 0.07 | 0.08 | 0.09 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 0.0 | 0.70 | 0.71 | 0.72 | 0.72 | 0.73 | 0.74 | 0.75 | 0.76 | 0.77 | 0.78 |
| 0.1 | 0.78 | 0.79 | 0.80 | 0.81 | 0.82 | 0.83 | 0.84 | 0.85 | 0.86 | 0.87 |
| 0.2 | 0.88 | 0.89 | 0.90 | 0.91 | 0.92 | 0.93 | 0.94 | 0.96 | 0.97 | 0.98 |
| 0.3 | 0.99 | 1.00 | 1.01 | 1.02 | 1.03 | 1.05 | 1.06 | 1.07 | 1.08 | 1.10 |
| 0.4 | 1.11 | 1.12 | 1.13 | 1.15 | 1.16 | 1.18 | 1.19 | 1.20 | 1.22 | 1.23 |
| 0.5 | 1.24 | 1.26 | 1.27 | 1.29 | 1.30 | 1.32 | 1.33 | 1.35 | 1.36 | 1.38 |
| 0.6 | 1.40 | 1.41 | 1.43 | 1.44 | 1.46 | 1.48 | 1.50 | 1.51 | 1.53 | 1.55 |
| 0.7 | 1.57 | 1.58 | 1.60 | 1.62 | 1.64 | 1.66 | 1.68 | 1.70 | 1.72 | 1.74 |
| 0.8 | 1.76 | 1.78 | 1.80 | 1.82 | 1.84 | 1.86 | 1.88 | 1.90 | 1.93 | 1.95 |
| 0.9 | 1.97 | 2.00 | 2.02 | 2.04 | 2.06 | 2.09 | 2.11 | 2.14 | 2.16 | 2.19 |


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[^1]:    * The requirements of Section 6 are illustrated in Fig. 1.

[^2]:    * Top refers to the position in a structure and not to the position of the specimen in the testing machine. This convention is used throughout this paper.

